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AMRL-TR-75-50 Volume 84



USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK

Volume 84

C-9A Aircraft, Near and Far-Field Noise

APRIL 1977



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AEROSPACE MEDICAL RESEARCH LABORATORY AEROSPACE MEDICAL DIVISION AIR FORCE SYSTEMS COMMAND WRIGHT-PATTERSON AIR FORCE BASE, OHIO 45433

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FOR THE COMMANDER

HENNING E. VON GIERKE

Director

Biodynamics and Bionics Division Aerospace Medical Research Laboratory

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with and without standard Air Force ear protectors. Far-field data measured at 19 locations are normalized to standard meteorological conditions and extrapolated from 75-8000 meters to derive sets of equal-value contours for these same seven acoustic measures as functions of angle and distances from the source. Refer to Volume 1 of this handbook, JUSAF Bioenvironmental Noise Data Handbook, Vol 1: Organization, Content and Application, AMRL-TR-75-50(1) 1975, for discussion of the objective and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing definitions of quantities, symbols, equations, applications, limitations, etc.

PREFACE

This report was prepared by the Biodynamic Environment Branch, Aerospace Medical Research Laboratory, under Project/Task 723104, Measurement and Prediction of Noise Environments of Air Force Operations.

The author gratefully acknowledges Mr. John Cole for his assistance in preparing this report, Mr. Robert Lee and Mr. Jerry Speakman for their assistance in acquiring the raw data, Mr. Keith Kettler, Mr. Henry Mohlman and Mr. David Eilerman of the University of Dayton for assistance in the mechanics of data processing, and Mrs. Norma Peachey and Mr. Mike Patterson for assistance in typing and preparation of the graphics.

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INTRODUCTION

The USAF C-9A is an aircraft that airlifts medical patients and is powered by two JT8D-9A turbofan engines. The aircraft was manufactured by the McDonnell Douglas Corp and the engines by Pratt and Whitney, a Division of United Aircraft.

This volume provides measured and extrapolated data defining bioacoustic environments produced by this aircraft during ground runup operations. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with ground runups of the C-9A aircraft.

This volume is one of a series published by the Aerospace Medical Research Laboratory (AMRL) under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew loations and in surrounding communities by operations of Air Force aircraft and ground support equipment. The far-field, community-type noise data in the handbook describe the noise produced during ground operations of aircraft, ground support equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discusses the objectives and design of the handbook, types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. Volume 2 provides a method and data for adjusting the handbook's far-field noise data, which are for standard meteorological conditions (15 C temperature, 70% relative humidity, 0.760 meters Hg barometric pressure), to derive comparable data for other meteorological conditions. Refer to Volumes 1 and 2 (references 1 and 2) for such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., inflight/flight crew and passenger noise, near-field/ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published. This index is periodically updated as individual volumes are published and is available upon request from AMRL/BBE, Wright-Patterson AFB, OH 45433. Organizations on the distribution list for the handbook will automatically receive a copy of each updated index.

Direct any questions concerning the technical data in this report and other handbook volumes to: AMRL/BBE, Wright-Patterson AFB, OH 45333; AUTOVON 78-53675 or 78-53664; Commercial (513) 255-3675 or (513) 255-3664.

Cole, John N., USAF Bioenvironmental Noise Data Handbook, Volume 1: Organization, Content and Application, AMRL-TR-75-50 (1) Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.

Cole, John N., USAF Bioenvironmental Noise Data Handbook, Volume 2: Procedure to Evaluate Effects of Non-standard Meteorological Conditions on Far-Field Noise, AMRL-TR-75-50 (2), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.

NEAR-FIELD NOISE

MEASUREMENTS

AMRL acquired near-field noise data on the C-9A aircraft during ground runup operations of its turbofan engines. For these tests the aircraft was located on a taxiway at Wright-Patterson AFB with no significant reflecting surfaces in the vicinity except the ground plane. Table 1 gives the surface meteorological conditions and the three engine/power conditions. The ground crew chief selected power conditions and near-field locations generally used during routine maintenance or engine runup for preflight checks.

At each near-field location a test engineer randomly moved a hand-held microphone in and around each location, probing all areas where a crewmember's head would normally be located. He recorded all of the noise samples on magnetic tape. During analysis of each sample, he determined the root-mean-square sound pressure using a 4- or 8-second integration time to derive a power-averaged level for each location. Figure 1 shows the four near-field locations where ground crew are usually located for maintenance and/or preflight checkout operations. Estimates of noise levels at other locations in the near-field are difficult since the noise source is spatially distributed, i.e., not a point source. The noise levels at near-field locations can vary widely depending upon relative distances from each noise source (intake noise, exhaust noise, panel resonances, internal engine noise through the engine wall, etc.).

Table 1 lists the numeric/alphabetic designators used on the data pages in this report to identify the measurement locations and test conditions. For example, the designator 1/A means ground crew location 1 and test condition A.

RESULTS

The measured data presented in Table 2 define the sound pressure levels (SPL) produced by the C-9A aircraft at the four ground crew locations. This table includes the overall, 1/3 octave band, and octave band levels. From these data one can calculate the variety of measures given in Table 3, which are widely used to assess the effects of noise on personnel and their performance.

All near-field data are for the meteorological conditions at time of test but are valid for all typical airbase meteorology because of the short sound propagation distances involved.

TABLE 1

MEASUREMENT LOCATIONS AND TEST CONDITIONS FOR NEAR-FIELD NOISE MEASUREMENTS

C-9A Aircraft, Ground Runup, Wright-Patterson AFB 8 October 1974, Tail # 10958

Ground Crew Locations

- Direction

1	Fire Guard
2	Marshal
3	Telephone Talker
4	Trim Adjustment
Aircraft Engine Operation	
A	Both Engines Idle
B	Both Engines 1.8 EPR
c	Both Engines 1.9 EPR
Meteorology	
Temperature	13.3 C
Bar Pressure	0.742 M Hg
Rel Humidity	47 %
Wind — Speed	1.5 M/Sec (3 Kts)



.240 Deg

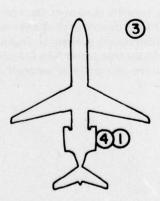


Figure 1. Near-Field Measurement Locations on a Taxiway at Wright-Patterson AFB, OH

FAR-FIELD NOISE

MEASUREMENTS

AMRL acquired near and far-field during a 1-hour test period, thus keeping similar meteorological conditions. Figure 2 shows the ground runup area, ground cover, aircraft orientation and the 19 microphone measurement sites on the semicircle. The center of the 75 meter radius semicircle used in surveying the JT8D-9A engines was on the ground directly below the intersection of the aircraft's centerline and the plane passing through engines' exhaust-nozzle exits. The ground runup pad did not have a blast deflector; therefore, the jets' exhausts were in a "free-flow" condition.

Table 4 provides cockpit readouts of some engine characteristics (EPR, fuel flow, etc.) for each power setting used in the far-field tests. Also listed in this table are the surface meteorological conditions during data acquisition.

All microphone measurement sites are in the acoustic far-field of the source where the sound wavefronts spherically diverge and the noise source may be regarded as a point source.

A portable microphone/tape-recorder system was used to sequentially record the noise at each far-field location. The microphone was attached to a hand-held pole, pointed at the source (0° angle of incidence) and vertically scanned from 0.5 to 3 meters for a period of 5-10 seconds during data acquisition at each microphone location. These samples were then time-integrated to derive a root-mean-square sound pressure level. Vertical scanning and time-integrating together reduce anomalies frequently present in data acquired by a fixed height microphone.

RESULTS

Table 5 lists the overall and 1/3 octave band SPL measured at the far-field locations under meteorological conditions at the time of the test. Data in all other figures and tables are based on these levels. These data were normalized to 100 meters distance and standard meteorological conditions (15 C temperature, 70% relative humidity, 0.760 meter Hg barometric pressure) and used to derive the graphic data in Figure 3 which provides a compact summary of the far-field noise characteristics of the C-9A aircraft in a standard format.

Figure 4 and Table 6 present two basic acoustic measures, the acoustic power levels and the directivity index, respectively. The acoustic power level describes the power radiated by the source as a function of frequency. The directivity index is a standard acoustical engineering measure that describes the geometric way in which the source radiates this power as a function of both frequency and angle from source. These basic source measures are primarily of interest for acoustical engineers and noise generation/control specialists.

Estimates of noise levels for intermediate power conditions (e.g., 1.75 EPR) and/or different number of engines operating (e.g., single engine) can be determined as explained in Volume 1 of this handbook.

Figures 5 through 11 are sets of equal noise contours describing seven different measures of noise as a function of angle and distance from the source for standard day meteorology. They are respectively, overall sound pressure level, C-weighted sound level, A-weighted sound level, perceived noise level, speech interference level, permissible exposure times for personnel and octave band sound pressure levels.

Data excessively influenced by spurious background/electronic noise were eliminated from all figures and tables. No data are presented at the 180 degree location for the 1.8 and 2.0 EPR power settings and at the 170, and 180 degree locations for the 1.7 EPR power setting because of turbulent air flow behind the aircraft. Typically, the A-weighted levels for these angles are 10 to 20 dBA below the level measured at the preceding microphone location.

Test personnel performed noise surveys during quiet periods when the background noise was minimal, e.g., early in the morning when no other aircraft or engine test stands were operating. Data eliminated because they were near the background/electronic noise were generally not significant because the levels were so low (e.g., Table 5 and Figure 11 at idle power).

Volume 2 of the handbook describes the influence of meteorology on far-field noise environments, and provides, if required, the factors necessary to adjust the handbook's standard meteorological day data.

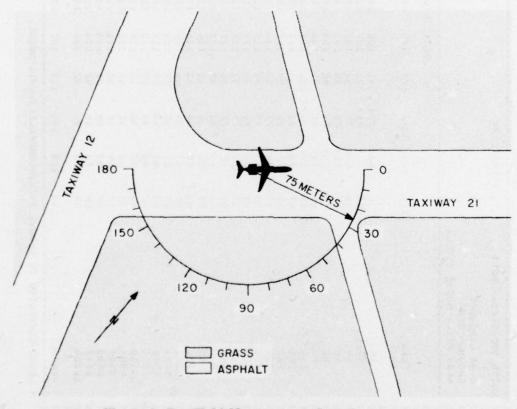


Figure 2. Far-Field Measurement Locations on a Taxiway at Wright-Patterson AFB, OH

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NOISE SOURCE/SUBJECT:	3.	OFERATIONS) AUN 01
C-9A AIRCRAFT GROUND CREW NEAR FIELD NOISE LEVELS) 12 MAR 76) PAGE F1
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25	96	65	704	91	102	102	
31.5	12		744	95	102	103	
0,	90	>69	7.0	06	102	103	
50	87	714	794	68	102	102	
9.3	6.8	144	7.3	88	103	104	
0.0	90	144	91	91	106	106	
100	92	754	9.0	100	103	109	
52	20	2,	90	53	113	112	
100	7 7	2.3	81	*	115	116	
250	0		22	25	115	117	
315	31	16	73	3	1117	118	
004	99	70	75	93	115	117	
200	9.0	72	11	96	117	119	
530	88	75	83	95	117	117	
900	88	14	40	35	116	117	
1000	99	82	84	63	116	117	
1250	99	99	80	16	116	116	
1600	93	87	90	26	114	115	
2000	30	83	93	68	112	115	
2500	91	93	66	16	113	113	
3150	93	26	96	26	115	115	
0000	31	88	91	26	112	114	
2000	19	40	18	95	111	111	
6300	29	83	88	35	110	109	
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FREU	1/A	2/4	3/8	4/4	4/3	1,0	
(42)							
31.5	35	7.1	80	96	106	107	
63	93	7.8	70	36	103	109	
125	35	82	99	101	116	110	
250	76	7.3	81	26	120	122	
500	34	8.0	40	96	121	122	
1000	33	69	30	96	120	121	
2000	96	15	100	101	118	119	
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				120	121						
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	111	1119	122	140	140						

¹⁰

TABLE 4

TEST CONDITIONS FOR FAR-FIELD NOISE MEASUREMENTS

C-9A Aircraft, Ground Runups, Wright-Patterson AFB 8 October 1974 Tail # 10958

Aircraft Engine Operation

Idle **Both Engines**

1.05 EPR, Engine Pressure Ratio 375 C EGT, Exhaust Gas Temperature 1000 LBS/HR FF, Fuel Flow

Both Engines 1.7 EPR Engine Runup

1.7 EPR 460 C EGT 5800 LBS/HR FF

1.8 EPR Engine Runup **Both Engines**

1.8 EPR 480 C EGT 6600 LBS/HR FF

1.9 EPR Engine Runup **Both Engines**

2.0 EPR 510 C EGT 8000 LBS/HR FF

Meteorology

Temperature 13.3 C 0.742 M Hg Bar Pressure Rel Humidity

47 % 1.5 M/Sec (3 Kts) Wind - Speed

240 Deg - Direction

	DISTANCE =	BAND 75	METERS														ONEGA 1.4	GA 1.4	
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FREG								A	ANGLE	(DEGREES)	ESI								
(7)	•	10	20	30	0,	20	09	20	90	06	100	110	120	130	140	150	160	170	180
25	65 <		654	704	734	684	684	684	724	684	684	104	684	999	104	734	734	734	17
31.5			>69	714	734	704	>69	104	714	734	734	714	*69	714	734	74.5	744	734	73
0,	724	>19	734	754	18	734	734	734	734	734	734	734	144	164	164	164	18	164	73
20	724	734	154	164	164	164	734	714	>69	714	734	164	164	744	744	164	744	734	2
63	204	744	724	734	164	734	>69			144	754	734	734	×69	734	144	174	164	2
08	×69	>69	714	734	164	714	×69		>69	734	>69	77	734	744	734	724	794	164	2
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250	72	11	73	12	202	69	67	99	99	202	69	29	99	69	99	69	16	75	99
315	73	75	72	73	7.1	69	89	99	29	99	29	69	49	29	19	72	18	73	3
004	7.0	16	7.1	7.0	69	67	29	29	89	99	29	49	61	69	69	99	25	20	9
200	69	73	72	72	7.0	99	19	69	71	29	69	65	29	63	65	99	73	7.0	99
630	11	12	11	72	2	68	69	69	2	69	11	96	63	99	29	29	*	7.1	65
900	73	15	72	7.4	73	1	68	69	20	72	73	99	63	99	29	69	73	11	99
1000	75	4	16	11	15	72	7.1	202	2	73	73	67.	249	929	674	67.	73	69	3
1250	91	85	80	19	92	73	7.1	2	2	73	72	29	19	99	29	69	12	20	65
1600	90	82	82	83	90	11	25	73	2	28	25	20	29	11	72	2	15	12	69
2000	83	8	88	83	82	85	91	80	73	11	73	1	99	69	69	1	2	2	69
2500	90	89	68	81	*	91	8	*	62	8	92	*	2	69	1	72	92	12	9
3150	*	83	83	83	82	00	4	18	11		75	72	2	72	72	23	11	73	69
0004	82	83	96	84	83	00	28	92	1	62	92	23	2	72	72	75	16	73	67
2000	62	19	62	90	7.8	15	7.3	7.3	69	73	2	29	99	29	99	69	7	99	62
6300	11	64	11	11	:	73	20	1.1	99	2	69	99	63	99	99	89	69	99	61
0000	14	92	:	13	1	20	99	19	99	67	69	63	9	63	63	3	99	62	25
10000	20	11	69	69	29	99	63	63	19	66	96	25	24	96	25	21	28	24	20
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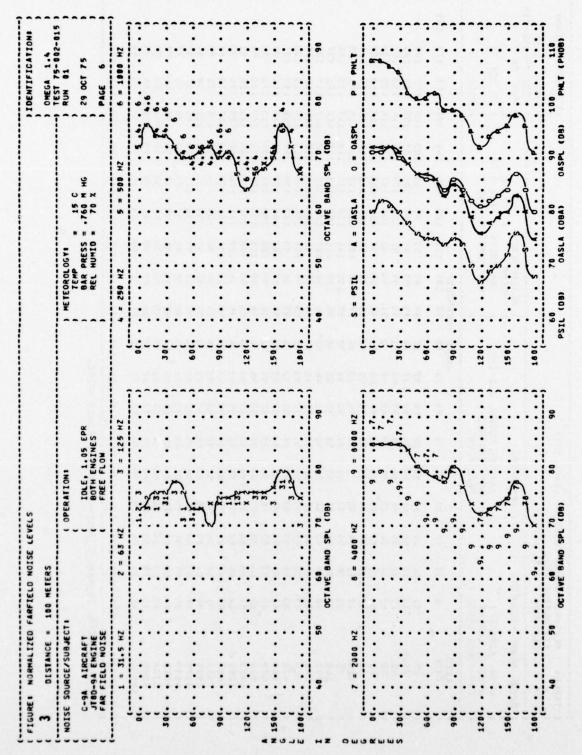
5	1/3 OCTAV	VE BAND	METERS	ERS													OMEGA	1.4	OMEGA 1.4
NOISE SOU	SOURCE/SUBJE	ECT		90	OPERATION:	. NO	100	102	18		ETEOR	METEOROLOGY	_ "				RUN	2 - 605	
C-9A A	AIRCRAFT				1.7 E	1.7 EPR ENGINE		RUNUR			BAR	PRESS		742 #	91	-	29 OCT	22	
FAR FIE	FAR FIELD NOISE				FREE	FREE FLOW	5				KEL	THOM					PAGE	2	
FREG								Ā	ANGLE	(DEGREES)	EESI								
(ZH)	•	10	20	30	3	20	6.0	20	90	90	100	110	120	130	140	150	160	170	180
52	7.8	79	90	80	81	81	82	98	95	85	85	87	91	96	100	103	103		
31.5			784	62	81	81	82	*0	98	87	88	89	92	16	103	105	105		
0,	82		83	82	18	98	87	85	68	87	88	36	16	100	101	108	101		
20	91	81	83	82	84	85	98	18	87	88	68	93	95	102	101	111	108		
63	91	83	*	94	92	85	87	98	87	90	06	95	96	103	109	113	111		
90	83	*	96	98	87	87	88	88	88	95	93	96	66	105	111	114	112		
100	96	88	87	87	87	88	06	06	91	16	95	96	101	101	115	116	115		
125	98	35	90	83	68	90	91	91	16	95	36	26	100	101	111	113	113		
160	93	93	95	90	96	90	95	91	66	36	16	96	101	106	113	114	114		
200	92	35	91	68	89	68	06	06	36	36	93	9	100	105	108	114	111		
250	92	66	95	68	87	18	80	80	06	91	36	66	86	102	101	112	101		
315	87	68	91	68	28	82	87	68	96	06	91	*	86	101	105	109	105		
9	10	10	6	90	9	*	92	00	50	2	60	55	2	-	103	101	701		
200	96	20	9	28	20	92	200	50	3 6	26	5	55	5	9	701	100	101		
200	20		.:	10			0 0	0 0	25	200	100	25		200	701	5			
	70			0	2 0	0	0 0		16	200	10	9 0		96	701	2	96		
1250	70		40	9 9		0 4	0 0	60	26	36	0 0	92	07	0	100	10	0 0		
1600		8		82		8			9.5	6	0	8	8	. 6		6	0 0		
2000	06		90	91	6	88	8	68	6	90	92	95	95	91	96	93			
2500	93		93	93	92	91	06	88	92	06	91	93	36	88	93		96		
3150	16		86	16	96	86	96	90	16	96	96	93	93	90	93	91	96		
0000	98		95	36	91	68	88	69	93	95	92	35	91	88	91	68	85		
2000	87		18	87	87	98	98	* 8	60	68	90	68	60	8	88	98	*		
6300	87	91	88	68	90	68	98	*	99	96	96	98	96	82	8 5	93	92		
8000	10		8	8	8	83	82	81	98	84	8	83	83	19	83	19			
10000	61	81	82	83	83	62	81	11	85	18	28	22	19	7.4	7.8	2	95		
OVERALL	102	103	104	103	103	103	103	102	105	105	106	108	111	115	120	123	121		

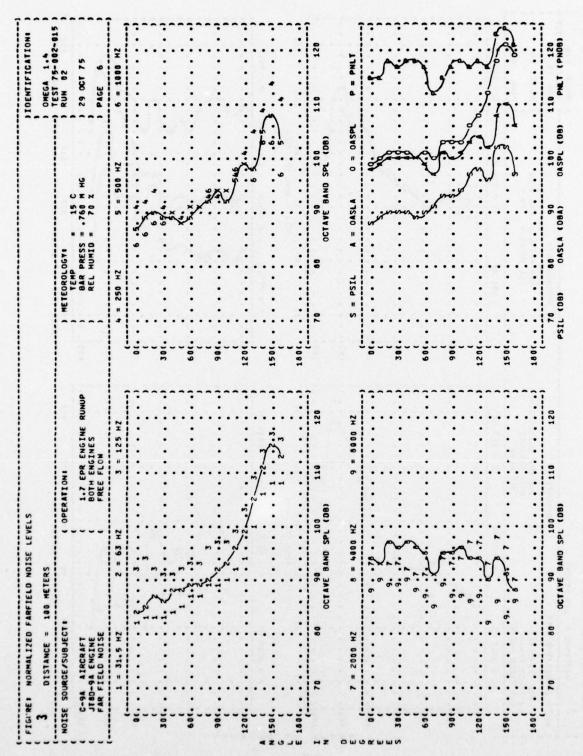
2	1/3 OCTAN DISTANCE		BAND 840	PRESSUR	RS	SOUND PRESSURE LEVEL E BAND = 75 METERS	(08)											OMEGA 1.4	1.4	a No
NOISE SC	SOURCE/SUBJE	DACAU	61.		90 1	OPERATIONS	. NO				ř	ETEOR(METEOROLOGY					RUN	03	
C-9A	AIRCRAFT	IFT				1.8 E	1.8 EPR ENGINE		RUNUA			BAR	PRESS		30	9		29 OCT	75	
FAR F	JT80-94 ENGINE	ISE				FREE	FLOW	ES				REL	HOMIO					PAGE	2	
FREG	e								Ā	ANGLE	(DEGREES)	EES)								
(HZ)		•	10	50	30	3	20	9	20	90	90	100	110	120	130	140	150	160	170	180
25		7.8	90	81	83	80	81	85	87	85	85	88	88	93	46	103	105	105	101	
31.5	.5	784	784	62	81	81	82	9 4	96	87	87	88	91	93	100	103	108	101	100	
;		62	81	*	94	96	98	96	88	87	68	90	36	95	102	106	111	112	102	
20		0.3	82	83	92	92	98	87	88	18	88	06	35	96	101	107	113	113	100	
63			80	4 1	98	85	87	80 6	200	60	91	93	*6	86	105	110	114	116	105	
98		* 6	4	0	9 8		00	5 0	96	2 6	26	* 6	900	102	104	112	115	121	1111	
125		06	93	95	90	88	96	95	93	93	95	97	66	102	107	113	116	119	101	
160		96	96	93	92	91	91	92	16	93	16	96	100	102	108	113	118	119	106	
200		91	93	91	91	68	96	91	91	91	76	96	96	103	101	109	118	116	101	
250		88	91	93	96	98	88	96	90	92	93	95	46	102	105	108	115	113	106	
315		90	91	93	91	68	68	8	90	91	36	16	95	101	104	106	113	111	102	
00,		90	68	06	96	8	98	69	69	69	90	6	93	101	102	103	110	110	86	
200		0	06	06	06	66	8	80	90	16	91	36	36	101	103	103	110	108	66	
630		90	0	60	06	86	200	06	91	91	5 6	96	16	101	103	103	109	106	*	
-		9 4		9.0				0 0	36	26	000	90	9 9	101	101	101	100	100	10	
1250		98	98	85	68	88	88	06	90	96	95	95	95	86	100	66	104	66	98	
1600		87	96	18	96	90	90	91	90	88	91	96	93	96	96	86	102	96	85	
2000		86	88	90	92	68	68	9.0	90	98	90	96	91	96	46	96	66	96	82	
2500		93	96	06	93	95	91	36	91	87	90	93	88	96	76	93	95	95	19	
3150		96	96	66	100	101	96	93	95	95	93	96	69	16	76	91	95	06	80	
0000		93	96	91	93	93	91	95	36	69	91	96	96	06	93	89	93	68	7.8	
2000		68	98	85	88	85	98	85	87	96	90	93	83	88	89	87	69	85	73	
6300		89	92	87	90	87	87	9 2	98	82	82	89	80	92	95	*	96	82	7.1	
8000		*	82	83	98	83	85	81	82	2	93	98	75	80	81	81	81	7.8	99	
10000			00	80	93	2	62	2.8	13	23	16	8	69	75	14	75	75	73	61	

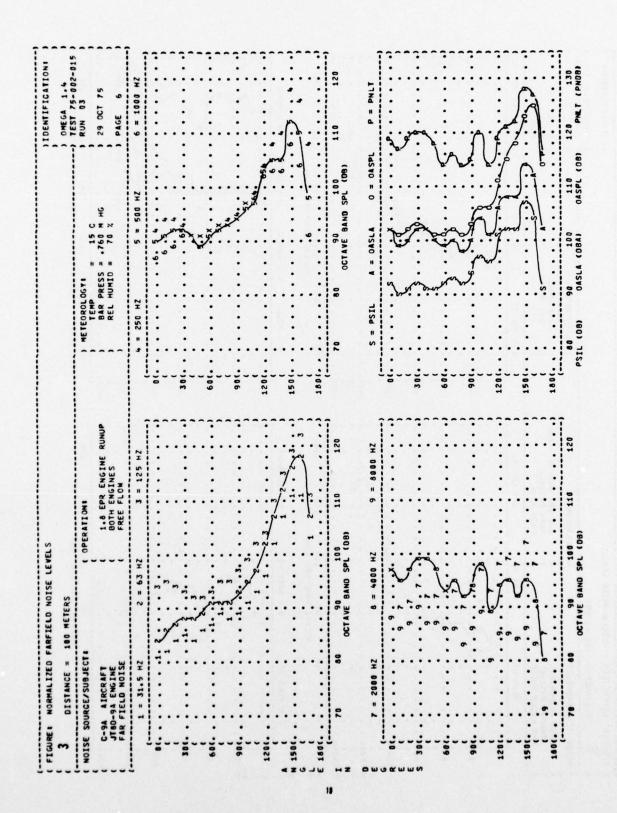
5 17	1/3 OCTAVE DISTANCE =		BAND 75 METERS	RS													ONEGA 1.4	1:	
NOISE SOURCE	SOURCE/SUBJEC	=		0	OPERATION	. NO					ME TEOROLOGY	0000	_ "				RUN	04	
C-94 AIRCRAFT	AIRCRAFT				TAKEO	TAKEOFF PONER,		2.0 E	EPR		BAR	PRESS		742 #	9		29 OCT	52 1	
FAR FIELD NOISE	NOISE				FREE	LON	2				7	1				-	PAGE	2	
FREG								1	ANGLE	(DEGREES)	EES)								
(HZ)	•	10	20	30	3	20	60	7.0	80	90	100	110	120	130	140	150	160	170	180
52	9.1	81	81	83	*6	96	86	87	87	89	89	93	95	66	105	107	107	103	
31.5	80	80	00	63	83	**	98	87	98	68	96	95	95	101	107	109	109	101	
04	82	83	9.6	84	84	88	68	88	16	9.0	*6	16	100	105	110	114	113	102	
20	82	48	96	85	85	89	69	90	90	91	46	96	100	101	112	116	115	103	
63	**	92	87	85	96	89	87	9.0	91	93	46	46	101	108	113	118	118	100	
90	10	88	60	90	60	90	91	92	92	96	16	66	105	110	116	119	119	111	
100	06	35	06	90	06	91	91	6	95	16	96	101	106	112	117	119	121	109	
125	45	5.0	32	56	15	56	56	36	26	16	100	101	105	111	118	119	120	100	
200		0 0	0 0	20	20	20	000	0 0	0.0	96	9 0	100	901	1:	116	121	118		
250	91	93	96	93	91	91	92	92	93	36	96	66	104	110	115	121	110	108	
315	93	93	96	93	91	91	92	6	92	66	86	100	104	109	113	119	118	106	
004	93	91	93	93	06	88	91	95	91	93	96	96	103	106	112	116	114	101	
200	99	91	91	91	90	68	95	93	92	96	96	100	103	101	111	115	112	96	
630	88	91	95	91	95	90	95	93	92	96	66	100	103	107	110	113	110	96	
900	96	88	06	35	95	95	93	16	*6	96	66	101	103	101	110	112	109	*6	
1000	10	88	9.8	06	06	90	95	93	93	96	66	100	102	101	109	111	106	91	
1250	**	98	88	68	90	90	91	93	16	96	66	100	101	106	107	109	104	06	
1600	92	87	06	90	91	90	93	93	93	96	66	100	101	105	106	107	103	88	
2000	91	91	06	06	35	06	93	93	*	96	66	96	100	104	103	105	100	83	
2500	06	90	68	90	68	88	91	93	3	16	86	46	66	101	101	102	86	91	
3150	88	68	88	90	90	89	95	92	36	95	86	66	96	101	100	100	16	80	
0004	60	90	91	91	91	69	92	91	95	96	96	66	26	66	96	96	95	11	
2000	**	82	85	9	84	9,6	98	87	91	93	*6	95	93	26	96	95	91	73	
6300	83	80	94	81	82	81	84	92	88	89	06	93	91	93	95	16	60	69	
0000	81	62	83	80	62	19	81	85	82	96	87	89	88	90	88	93	88	99	
00001	18	75	62	22	15	75	11	80	85	91	80	98	9.4	96	61	90	62.	09	
OVERALL	104	104	105	104	104	103	105	106	107	108	111	113	116	121	126	130	129	118	

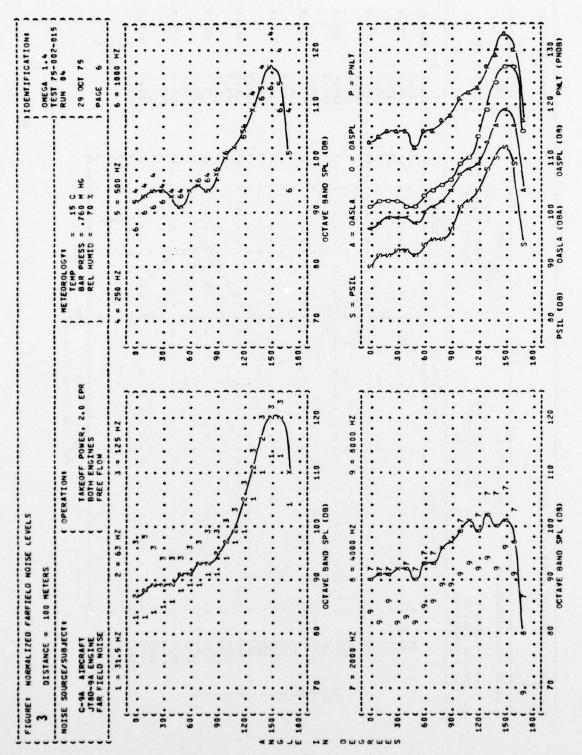
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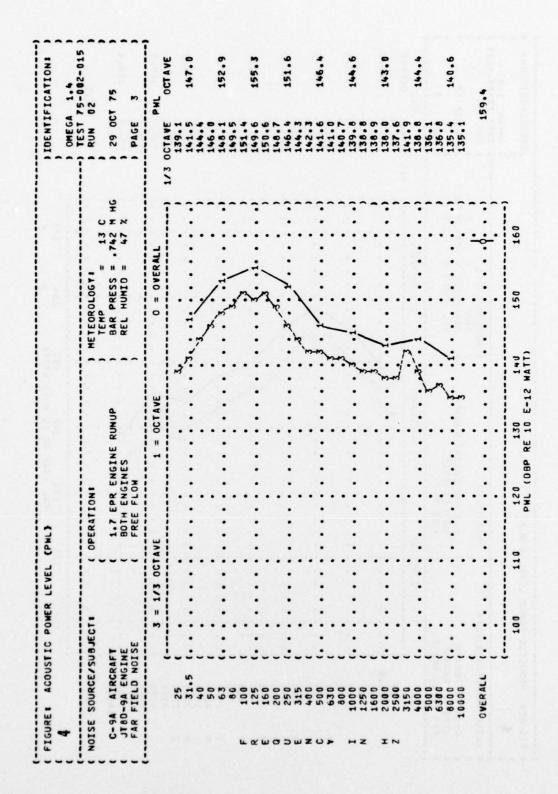


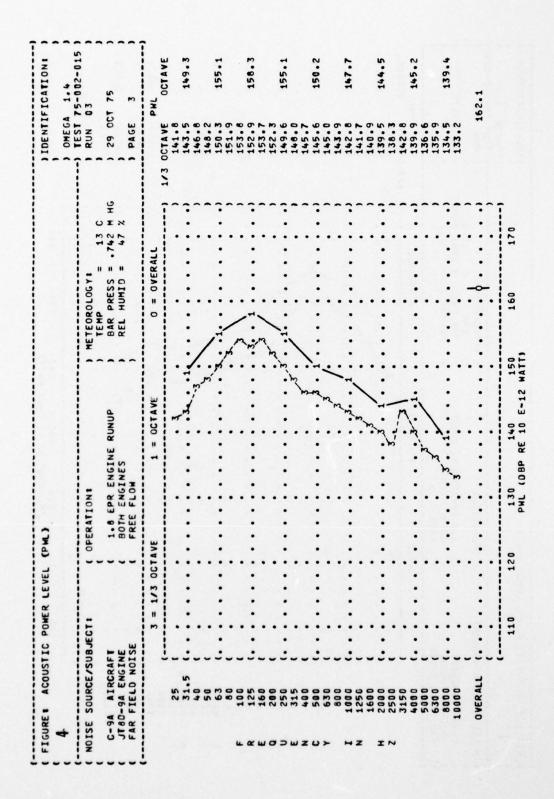


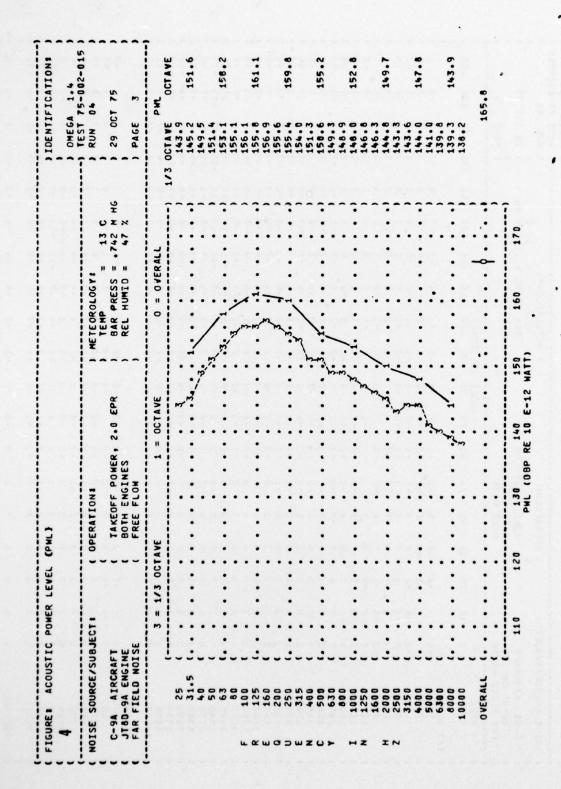




							OMEGA	OMEGA 1.4
NOISE SOURCE/SUBJEC	SUBJECT	COPERA	OPERATION		METEOROLOGY:		RUN	01
-94 AIRCR	MET	100	IDLE, 1.05 EPR	• • •	RESS		1 29 OCT 75	75
JIBD-94 ENGINE FAR FIELD NOISE	INE IOI SE	FRE	FREE FLOW		REL HUMID =	2 14) PAGE	P 0
	3 = 1/3	OCTAVE	1 = 00	OCTAVE	0 = OVERALL	11		PML
36				-			1/3 OCTAVE	OCTAVE
31.5		:		3	1	•	117.0	122.6
0 c				,,,		•••	119.9	
63						•	118.1	123.4
900		•		1	•	•	117.9	
125						-	117.4	122.1
160		•		1.5	•	•	118.3	
200		•	•		•	•	114.2	4.04
315				-			114.9	
004					•	-	113.4	
200	• • • • • • • • • • • • • • • • • • • •					•	116.1	116.8
900		•	• •	, 5			114.5	
1000				7.			117.5	122.4
1250				יית	. /	-	118.9	
1600				7.	1		122.2	
2000					· · · · · · · · · · · · · · · · · · ·	•	125.2	130.6
3150					1		125.4	
0004							126.6	129.8
2000					1	-	122.1	
6300		•		,	. '	•	121.4	
0000				· × · · · ·	. 15	:	120.5	125.1
1000				••			110.0	
OVERALL					•	•	13	135.2
	•	•	•	•	•	•		
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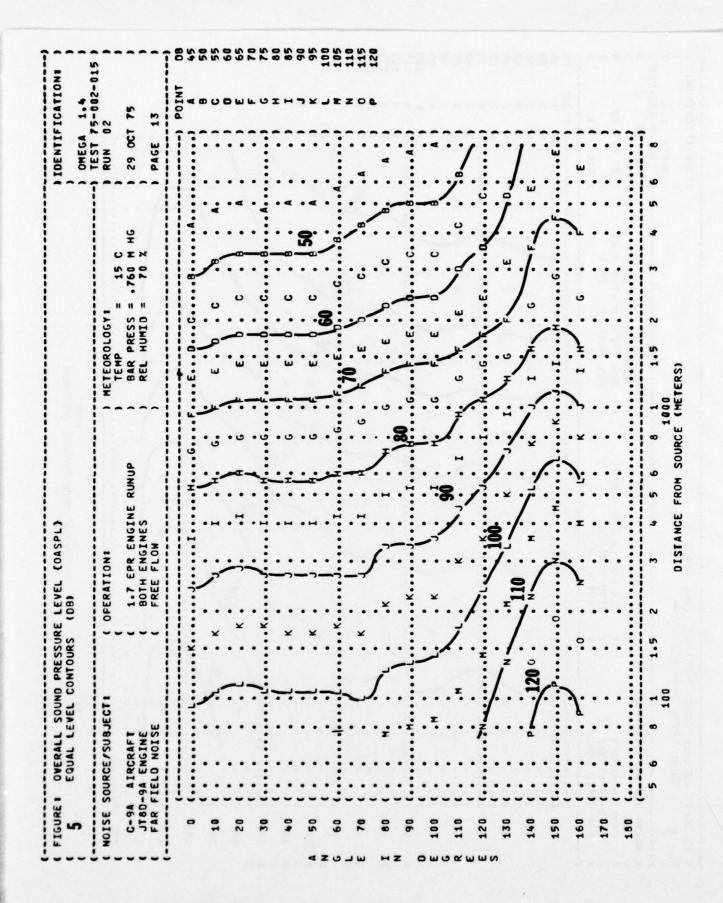
		-							-		-			-		-	TEST	75-002-015	2-01
NOISE SOURCE/SUBJEC	JECT			1 OPE	OPERATION	. N.				-	METEOROLOGY	0000				-	RUN	01	10-3
C-94 AIRCRAFT				_	OLE,	1.05					BAR P	PRES	","	13 C	94		29 OCT	1 75	
JT80-94 ENGINE FAR FIELD NOISE	·······································		-	80 LL	BOTH E	ENGINES	ES					HUMIO	11				PAGE	•	
FREG (HZ)	0	10	50	30	3	50	9	4 0	NGLE	(DEGREES) 90 100	EES)	110	120	130	1,0	150	160	Ë	180
1/3 OCTAVE																			
- 52	.5	•	5	?	m	-2	7	-5	~	-5	7	-	2.	-5	0	mı		m	٠,
31.5			7:	-	~ ~	? •		2 ?	7 1	- 7	~ :	7 ?	2 -		~ ~	m -	W W	~ -	
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63			17		m	. 0		,		-	0	. 0	-	*		, -	2		-
			-	-	*	-2	•		.3	0	-	7	-	2	1	?	9	4	-
		. 2	-1	0	*			7	7	-	7	-	•	7	-	-	•	m	7
125		2		7	+	-5	-5	-5	7	-	-	-	-	1	-	7	~	7	*
160		-	2		•	7		7	0		0	7	0	-	?	0	m .	7	? '
200	5		31	η.	?	?	m !	7		-	2.	2.	2.	7	٠.	?	•	? '	
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630	2	2	2	m	-	7	0-	0-	-	?	2	•	9	-3	-5	-5	8	2	*
900	2	2	~	m	m	•	-3	-5	?	7	m	*		-5		-5	m	0	.5
1000	~	~	*	· 2	m (0	7'	2.	2	~ (*		91			٠.	2	
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2000	2	9	2	9	*	1	?	-	†	?	-	9	6	9-	9-	-5		9	-12
6300	9	80	9	9	~	2	-	?		?	-2	-5	°	-5	-5	-3	-5	-5	-10
8000	9		9	2	m	~	-	-1	-5	7	-3	-5		•	-5	1	-5	9-	==
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OCTAVE																			
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125		-	-	0	~	-3		-5	?	•	-	•	•	-	-	0	m	0	1
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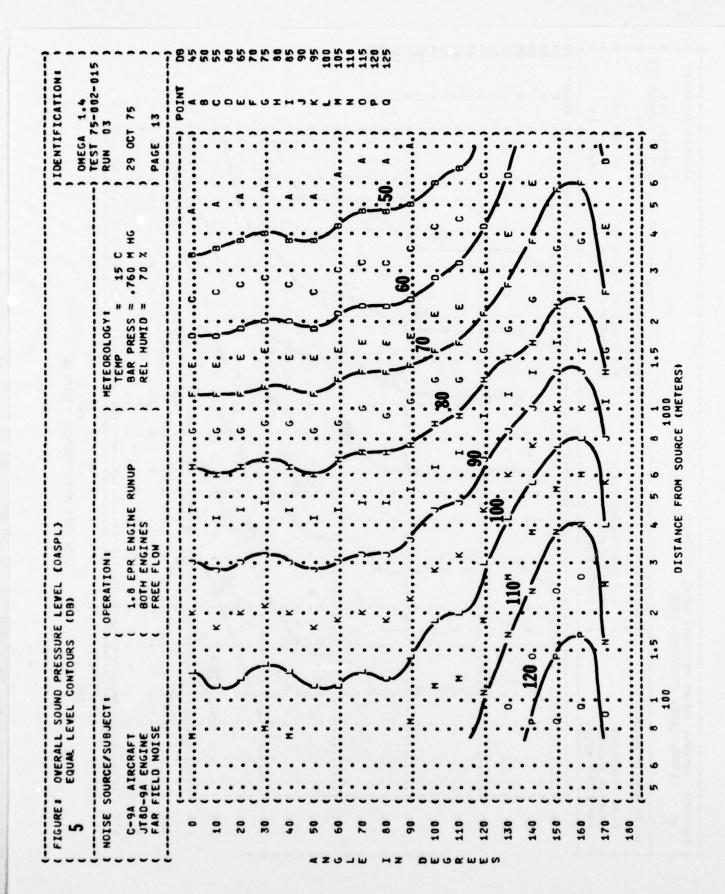
NOISE SOURCE																-	OMEGA	1:4	
	SOURCE/SUBJEC	113		100	OPERATION	I NO				-	ME TE OR OL OG Y	0106Y				1	RUN	02	-015
	-									~	TEMP	2000			,				
JT80-9A ENGINE	GINE				1.7 E	H ENGINES	ES	KONOP			REL	HUMIO	, ,	2 24	2		29 00	(2)	
FAR FIELD NOISE	NOISE			J	FREE	FLOW				-						-	PAGE	•	
FREG								•	ANGLE	(DEGREES)	EES)							!	
(142)	•	10	20	30	9	20	09	20	90	06	100	110	120	130	140	150	160	170	19
1/3 OCTAVE																			
52	-16	-15	-14	-14	-13	-13	-12	-8	6-	6-	6-		-3	-	•	6	6		
31.5	-19	-17	-18	-17	-15	-15	-14	-12	-11	-10	8		-5	-	1	σ	•		
0,	-17	-17	-17	-17	-15	-14	-15	-14	-10	-15	-11	•	-5	0	1	6	•		
20	-20	-20	-18	-18	-17	-16	-15	-14	-14	-15	-15		9-	-	9	=	1		
63	-22	-20	-19	-19	-18	-18	-16	-11	-16	-13	-13	-11		?	•	10	•		
90	-22	-21	-19	-18	-13	-17	-16	-16	-16	-12	-15					01	0 0		
100	02-	-18	-13	-13	-19	-19	-10	-16	-12	-15	-11	0			•	01			
165	97-	-13	*:	-10	110	*	-13	-13	01-	-10			•	2		,			
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007	-17	-15	21.	-12	-12	-12	114	-13	-15	71.	11.		* '	-	• •	::			
062	-10	-15		-15	1.	* .	-13	-13	11.	21.		0	5	H (0	7 .	۰ ۵		
515	-15	-10		-10	-15	**	21-	-10			0		7	V 6	0	2	o u		
200		200			100	2:	21-		9		2	17	•	7 7	0 4	90			
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1250	-10	-10	6-		9-	- 8	1	*	7	-2	•	~	*	-	9	8	-		
1600			9-	9-	-5	9.	-5	*	-5	-2	•	~	*	0-	2	*	-		
2000	-2	-2	-2	-	-	*	-3	.3	•	-2	0-	m	m	-	3	-	*		
2500	2	2	2	~	1	0	-	-3	-	-1	0	2	-	-3	2		-5		
3150	-	•	m	m	•	•	2	-5	?	•	-	7	7	-5	-5	1			
0004	-5	-5	-	-	-	-5	-2	-5	2	-	-	-	0	-3	-	-5	9		
2000	•	-5	-	9	0-	-5	-5	-3	7	-	2	-	-	1	-	-5	-3		
6300	0	2	2	~	~	2	7		~	7	7	7	-	- 2	7	*	-5		
8000	1	-	-	-	-	-	7	-2	2	-	-	?	7	*	-	1	-5		
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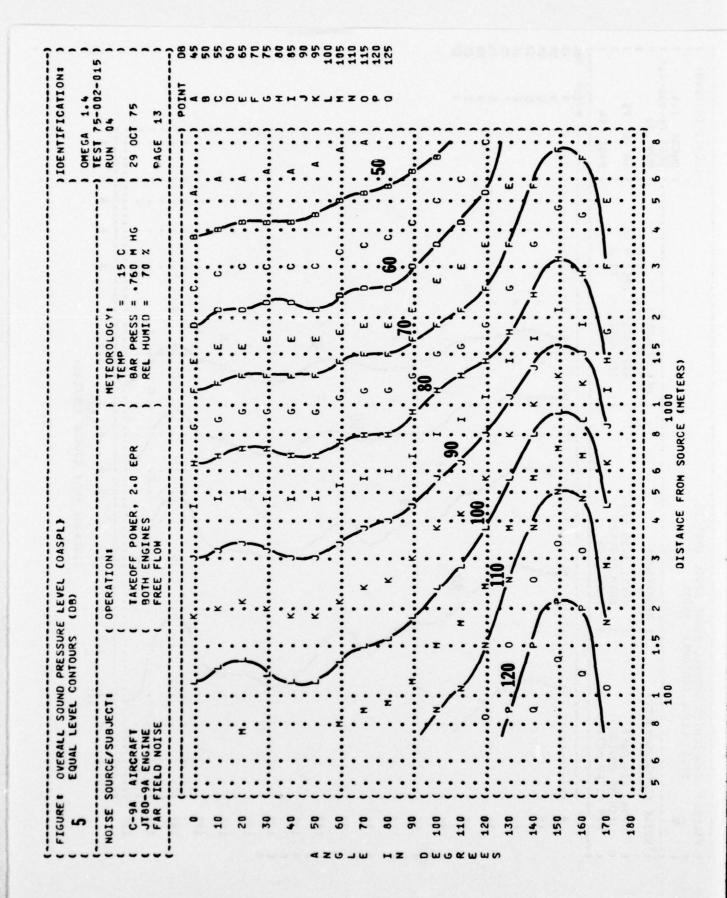
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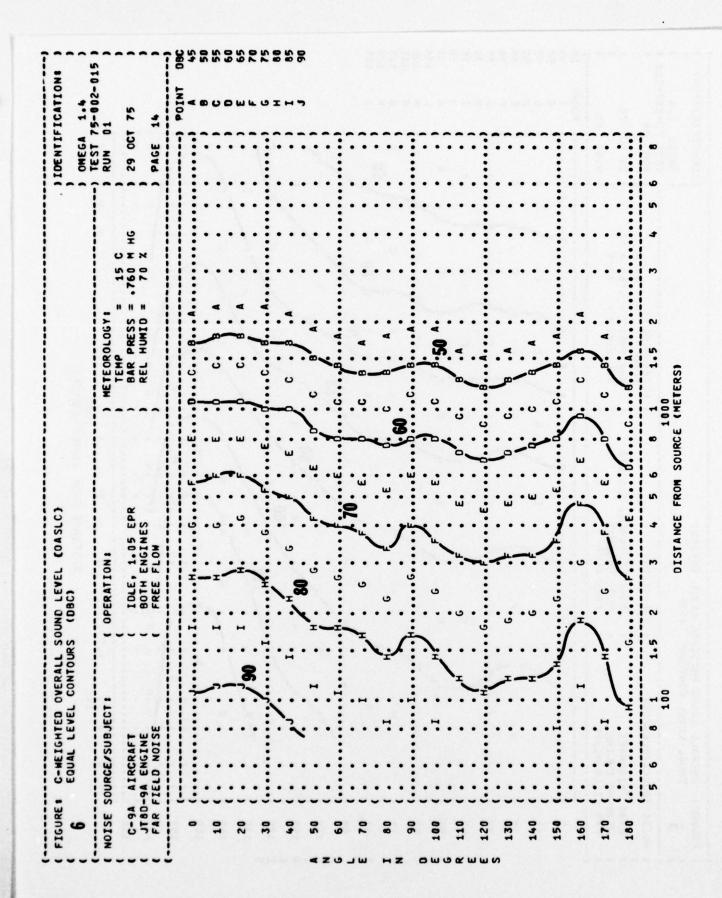
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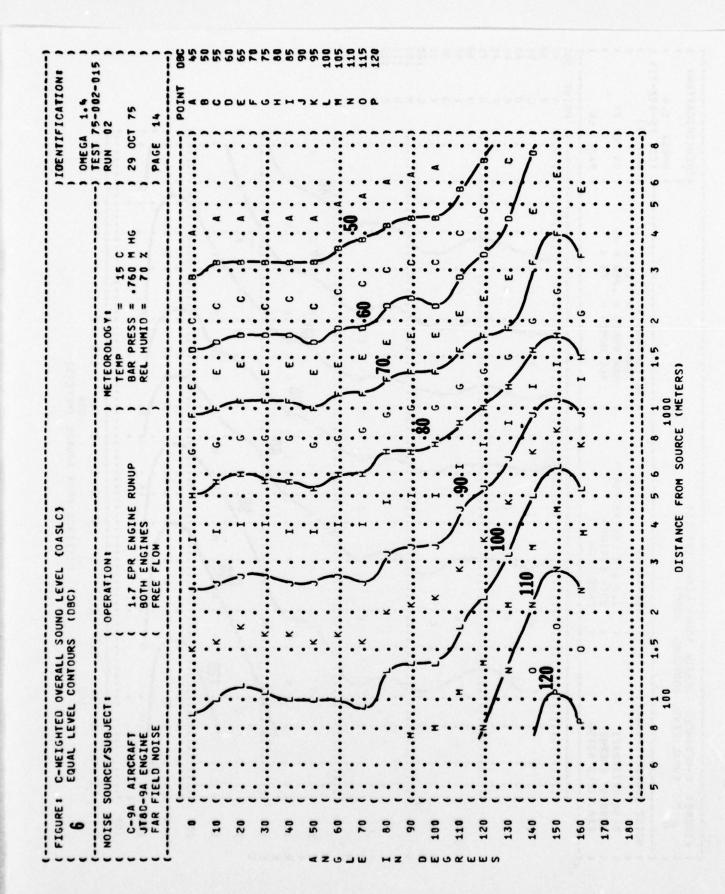
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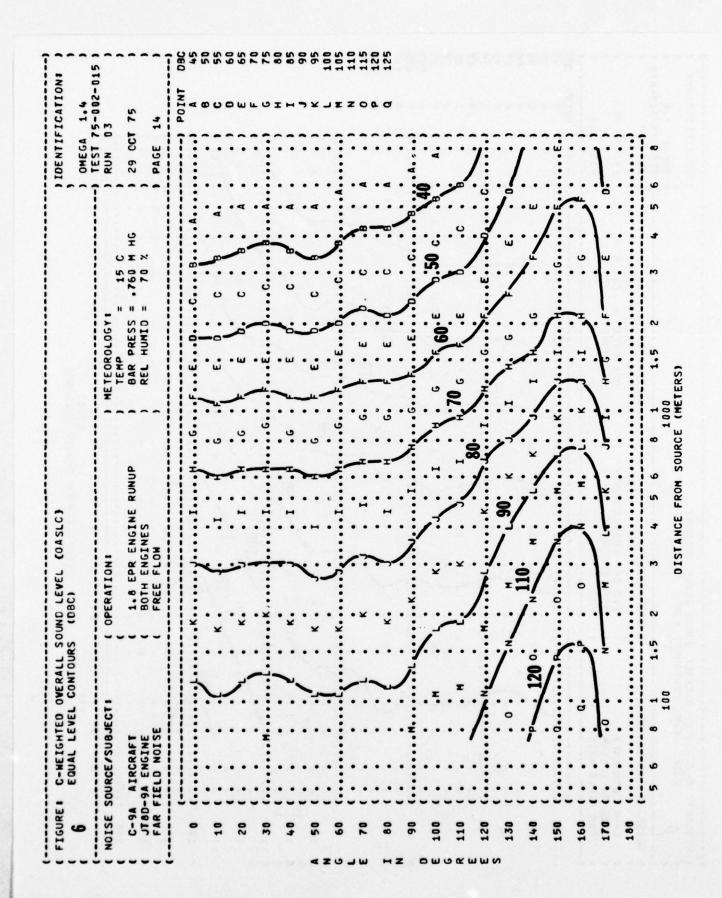


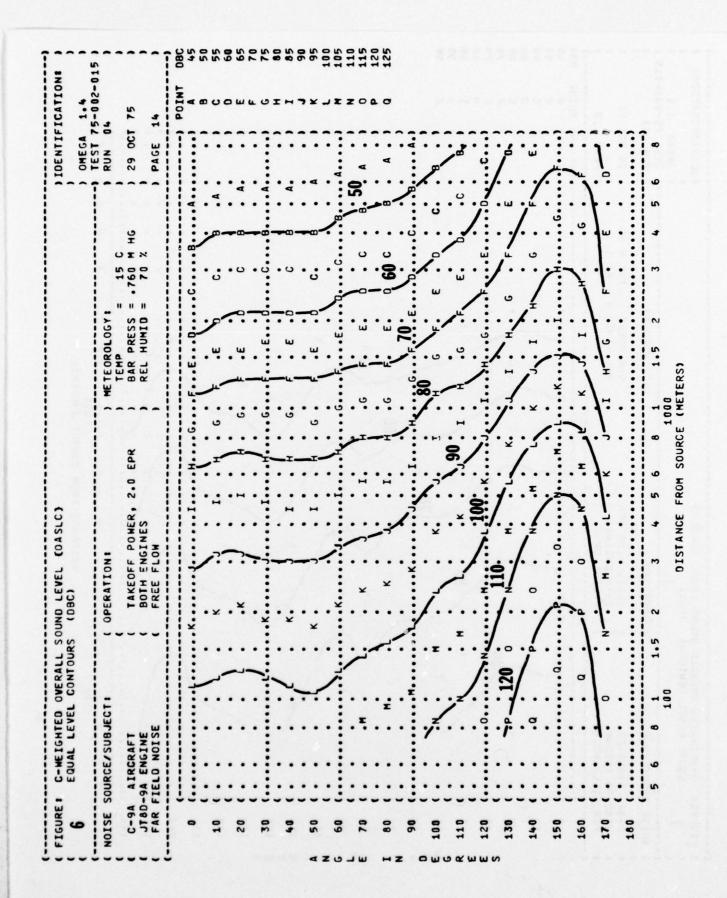


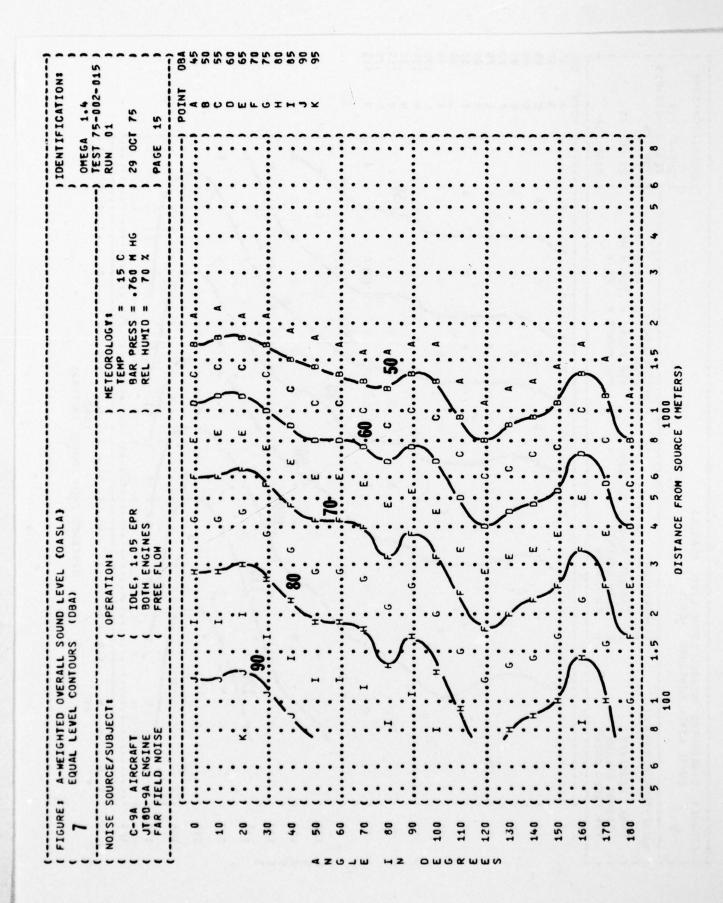


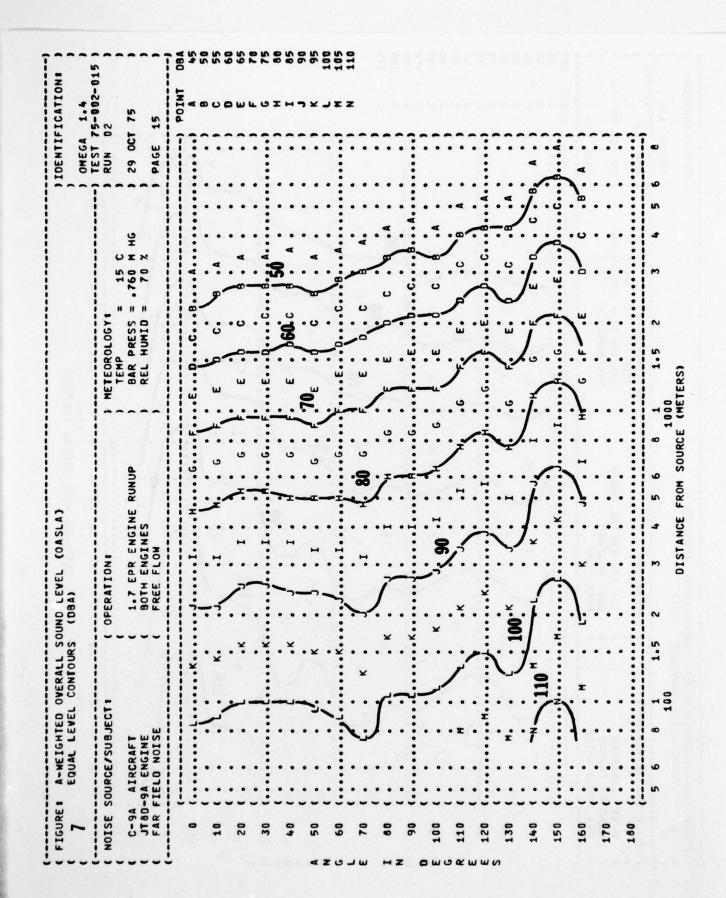




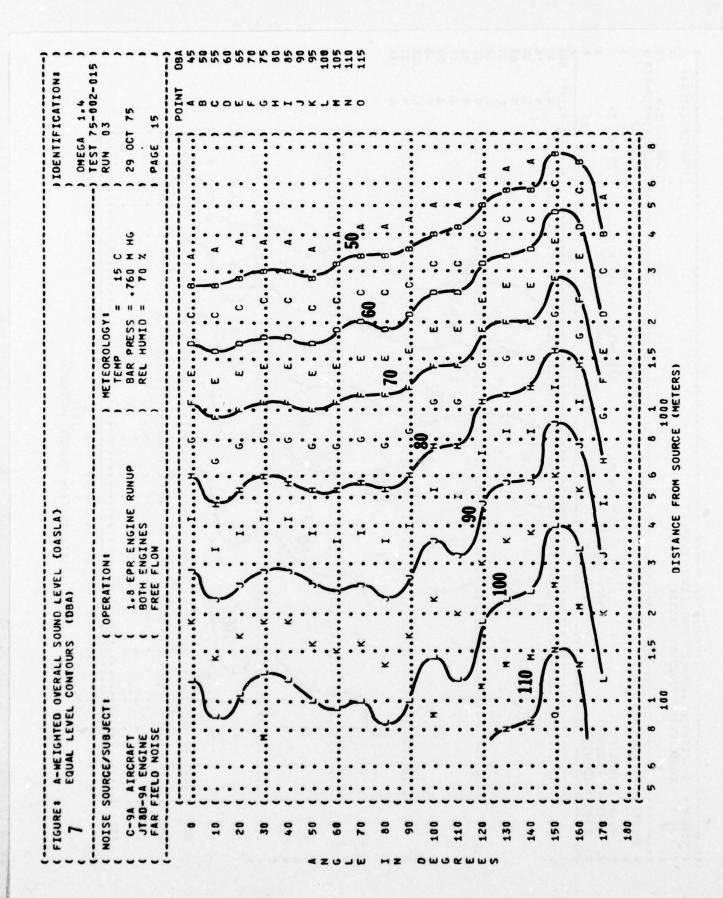


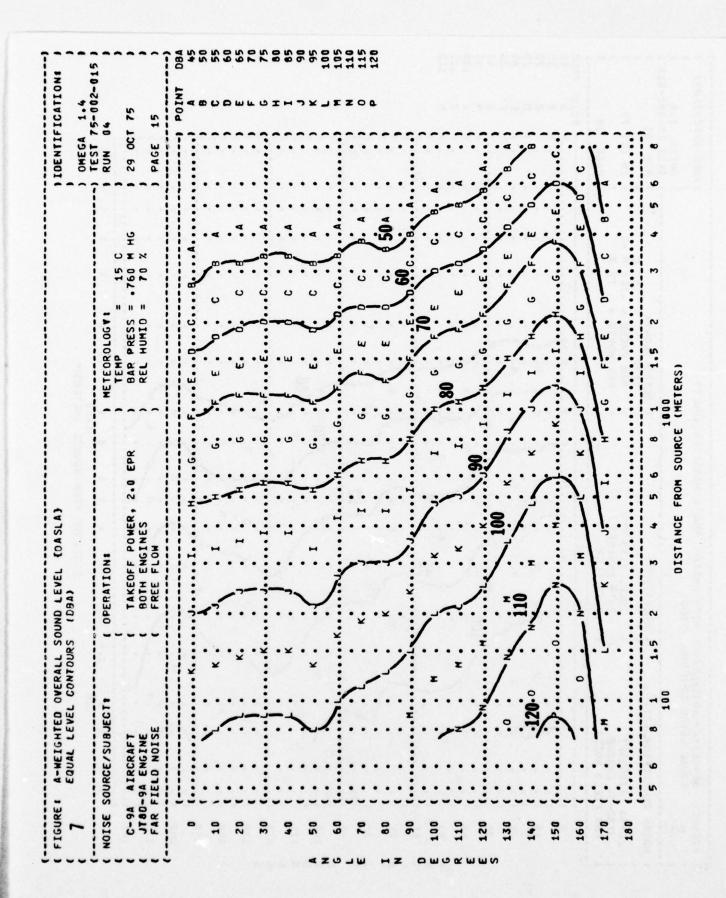




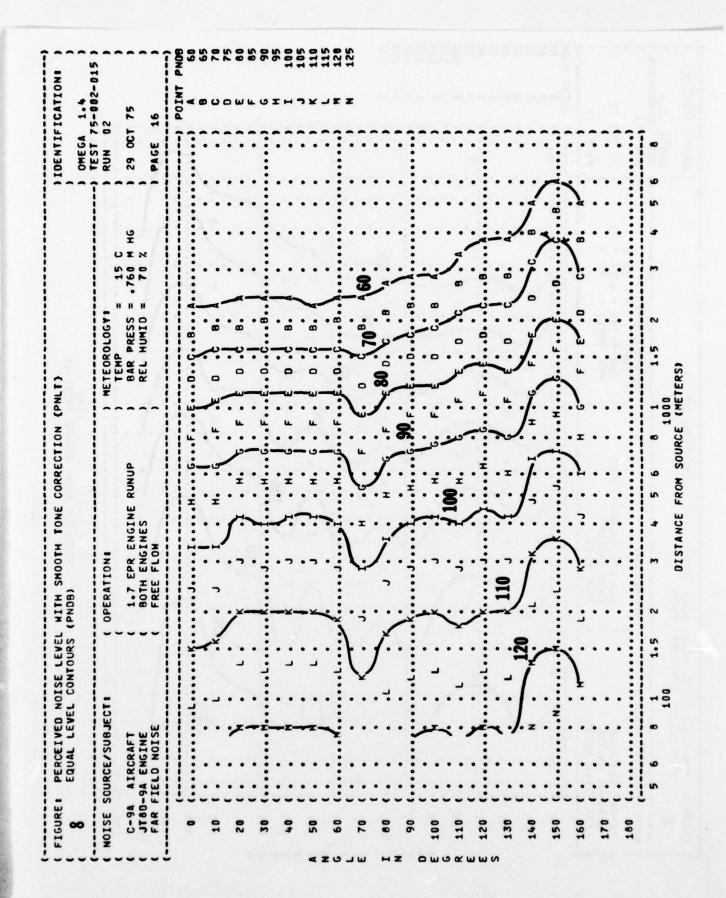


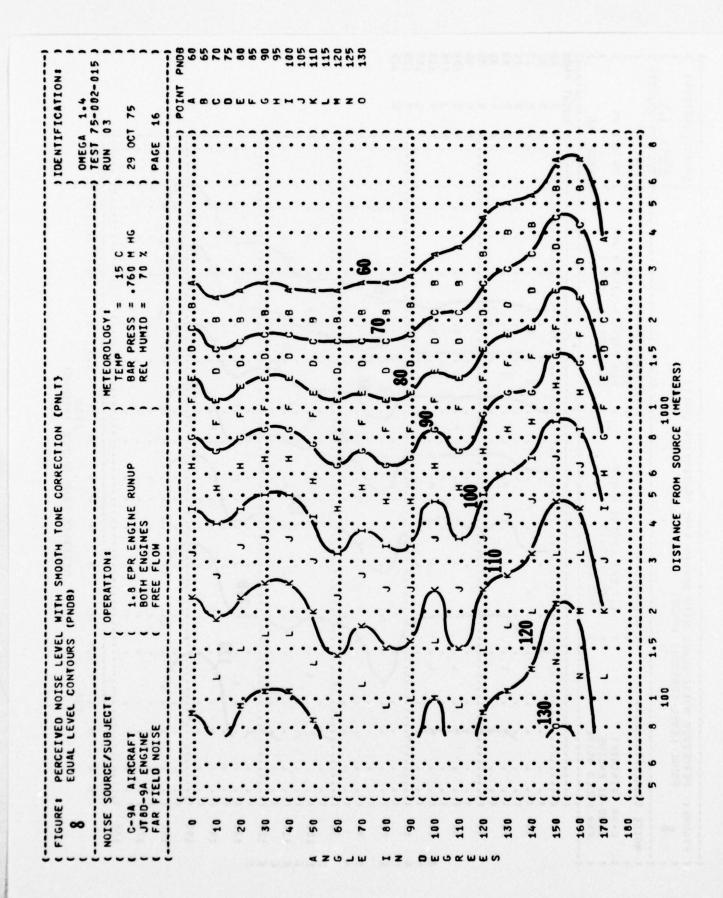
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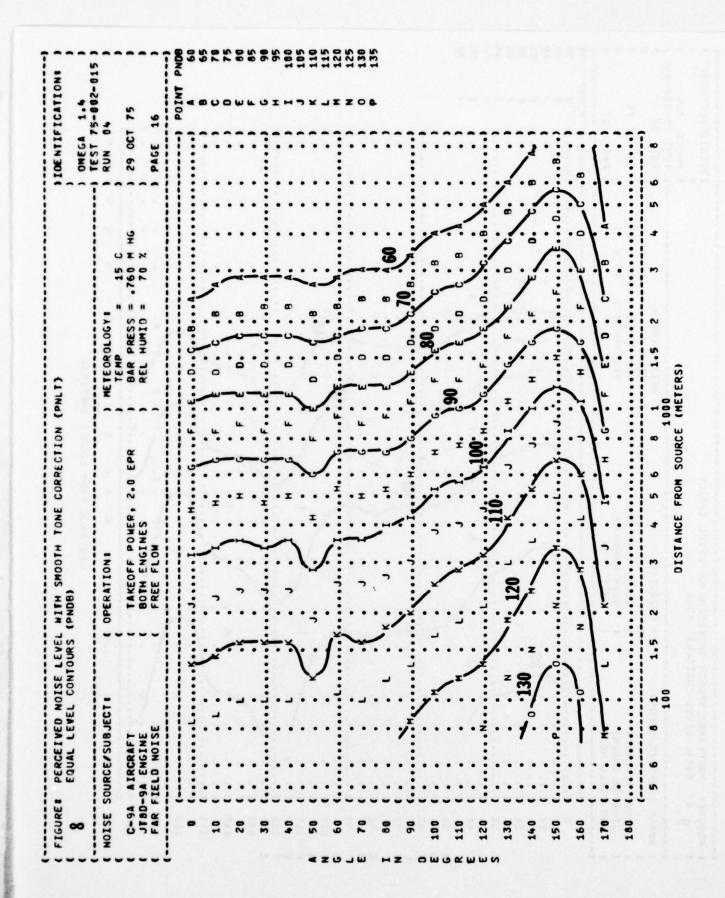




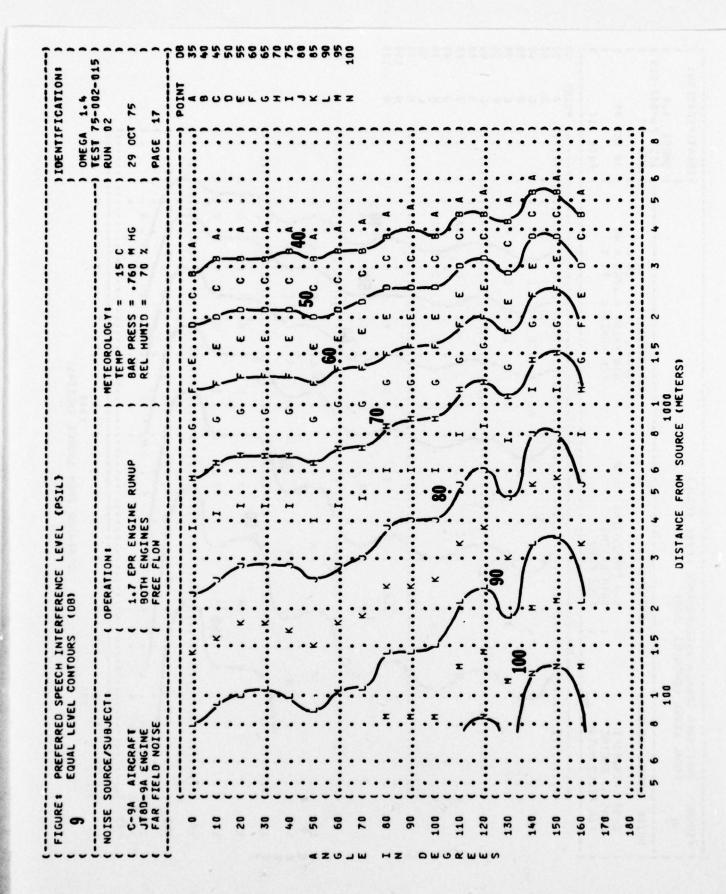
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FAR FIELD NOISE	(FREE FLOW) PAGE 16
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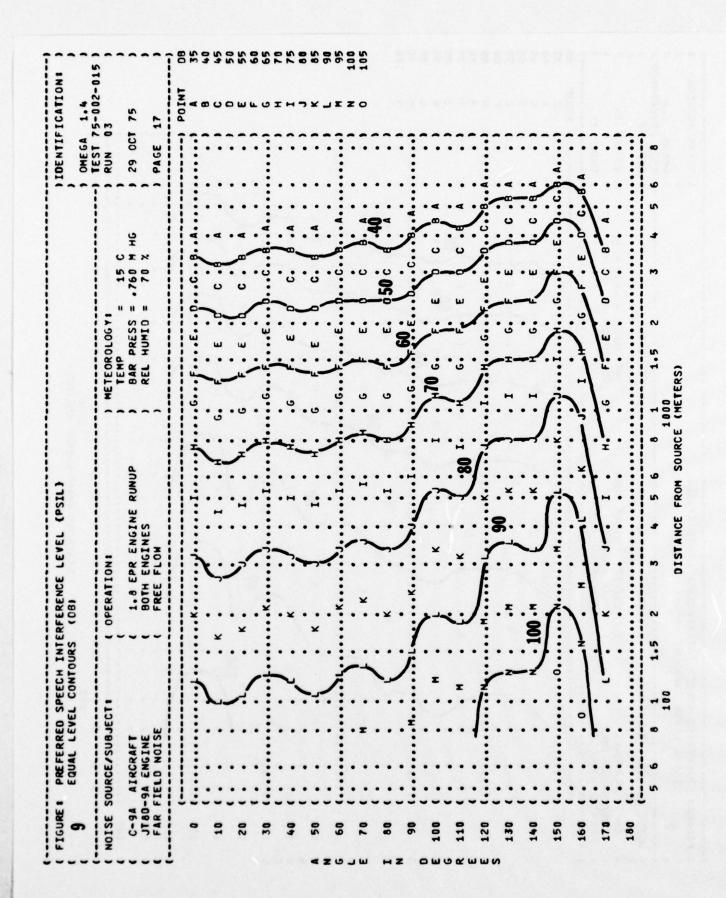


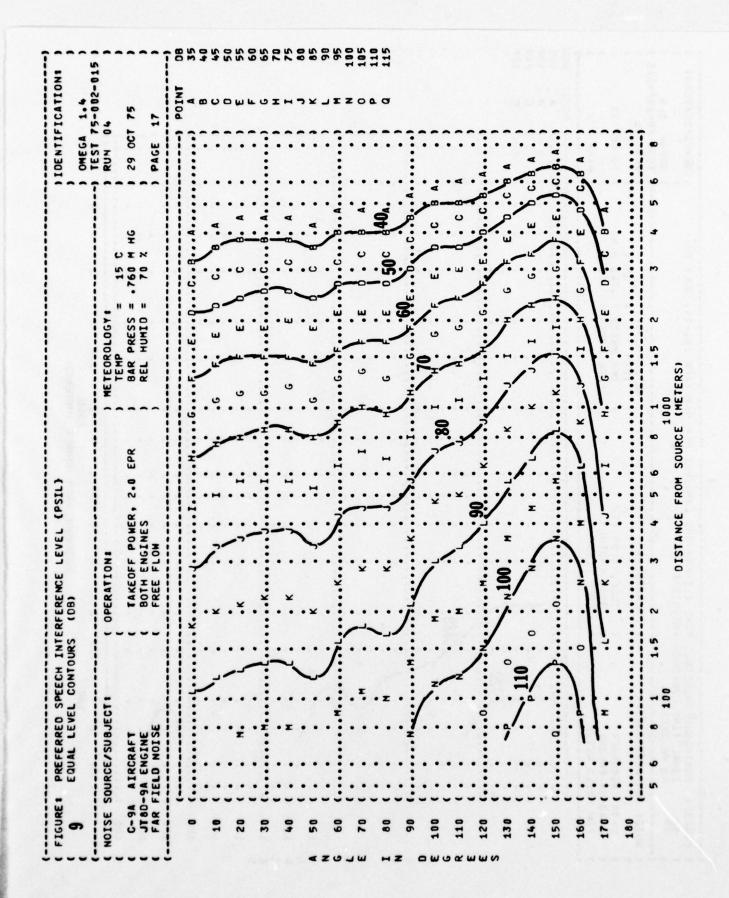




			(OPERATION:) METEOROLOGY: 15 C (IDLE, 1.05 EPR) BAR PRESS = .760 M HG (FREE FLOW)) METEOROLOGY:) TEMP = 15 C 05 EPR) BAR PRESS = .760 M INES) REL HUMIO = 70 X M
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E SOURCE (SUBJECT! (OPERATION!) HETEOROLOGY: 1.5 EPR ENGINE R FIELD NOISE R F	SOURCE/SUBJECTE AIRCRAFT -9A ENGINE FIELD NOISE (ATION:	*********						- [OMEGA	OMEGA 1.4
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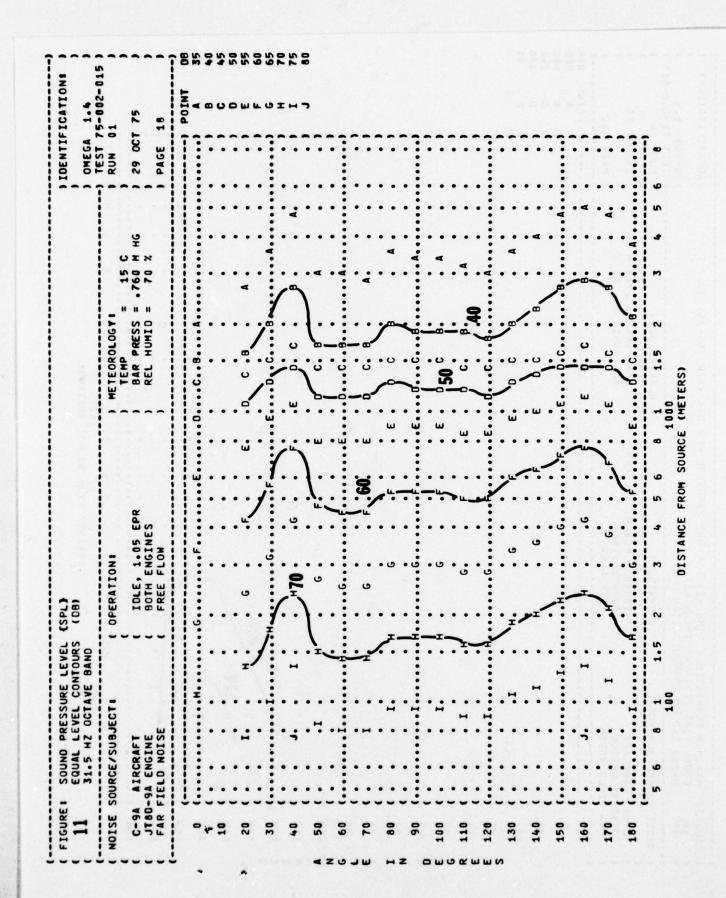
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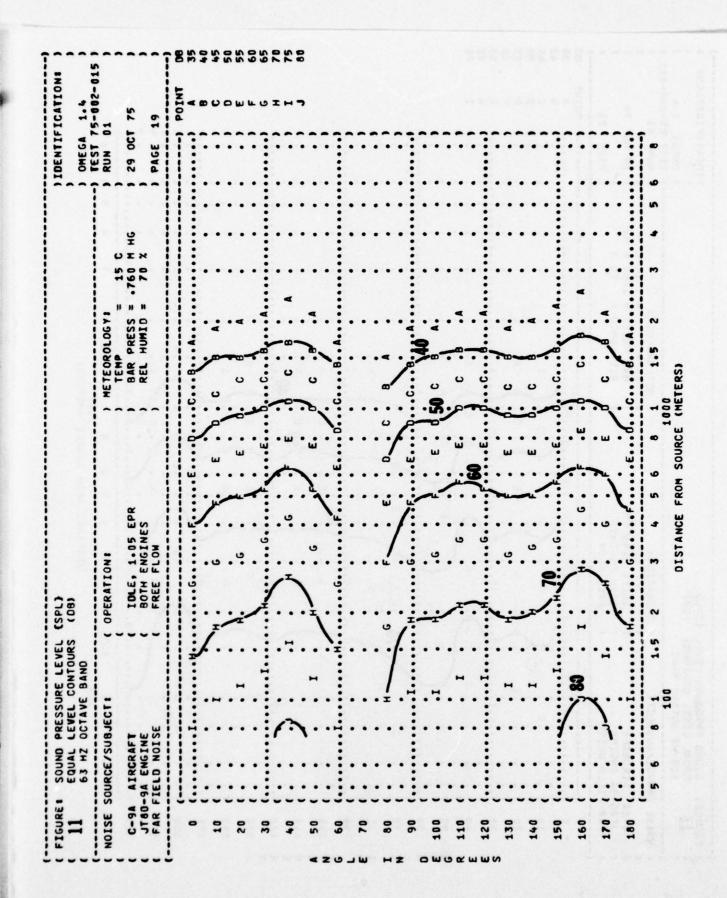
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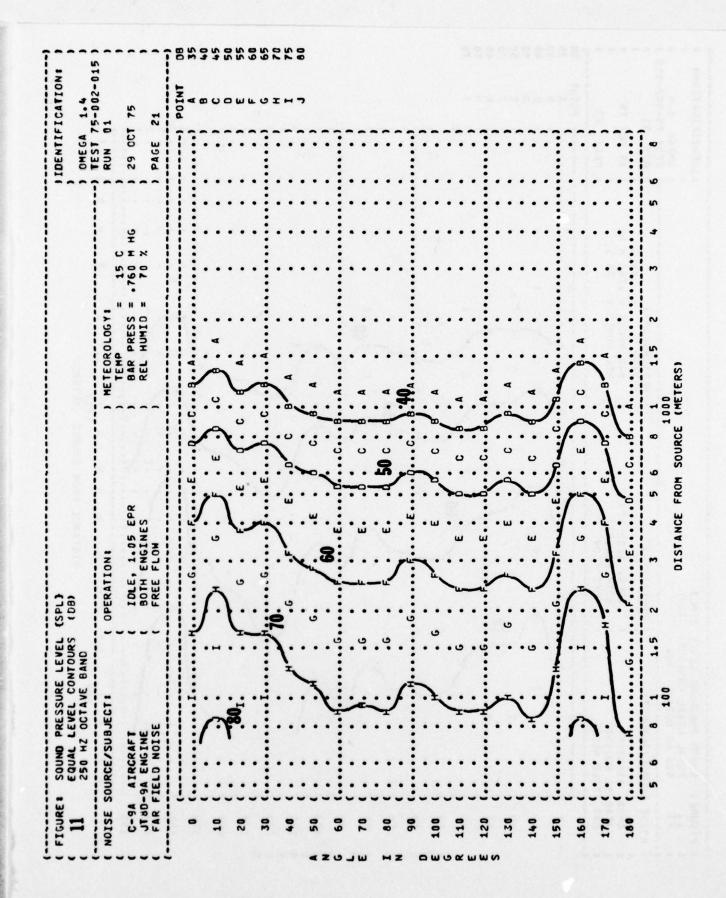
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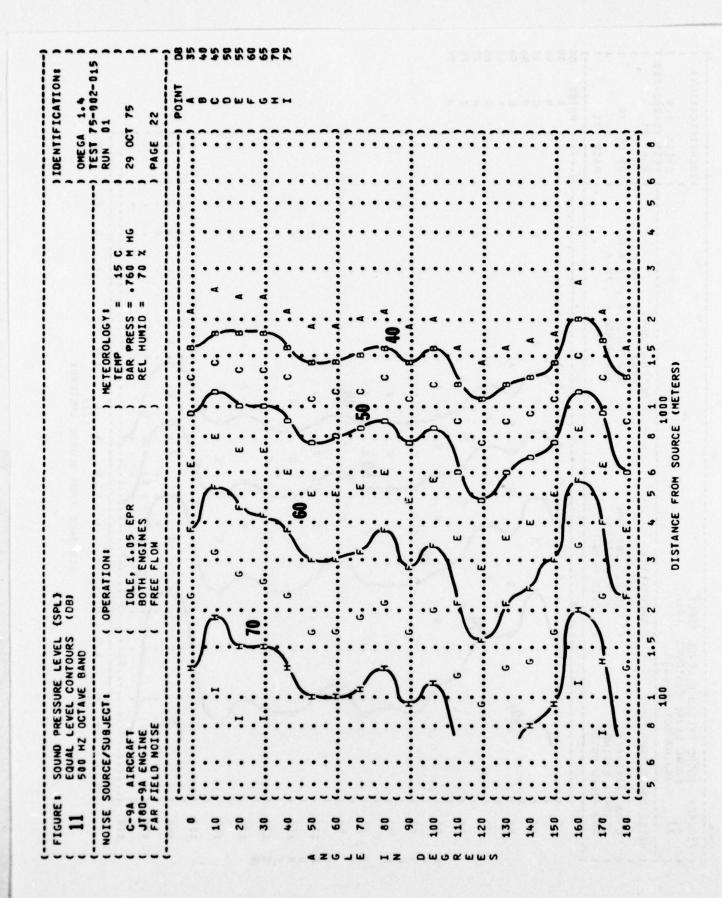
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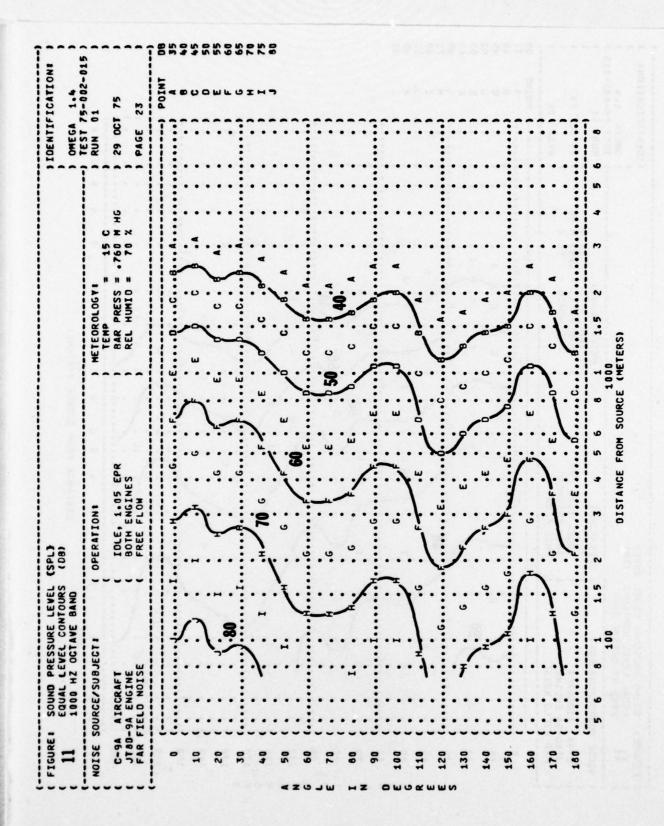


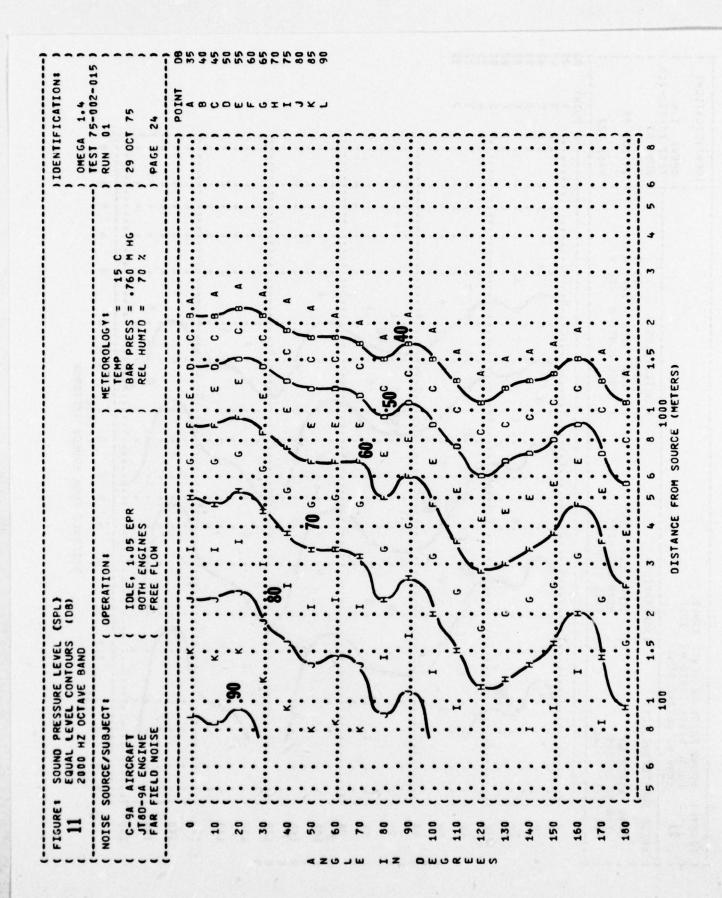


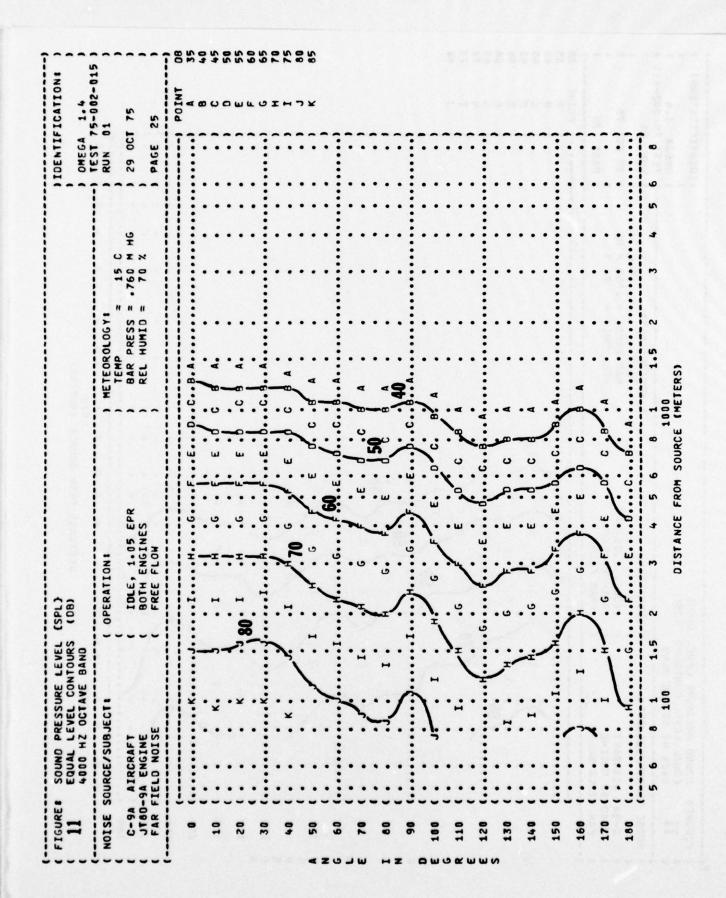
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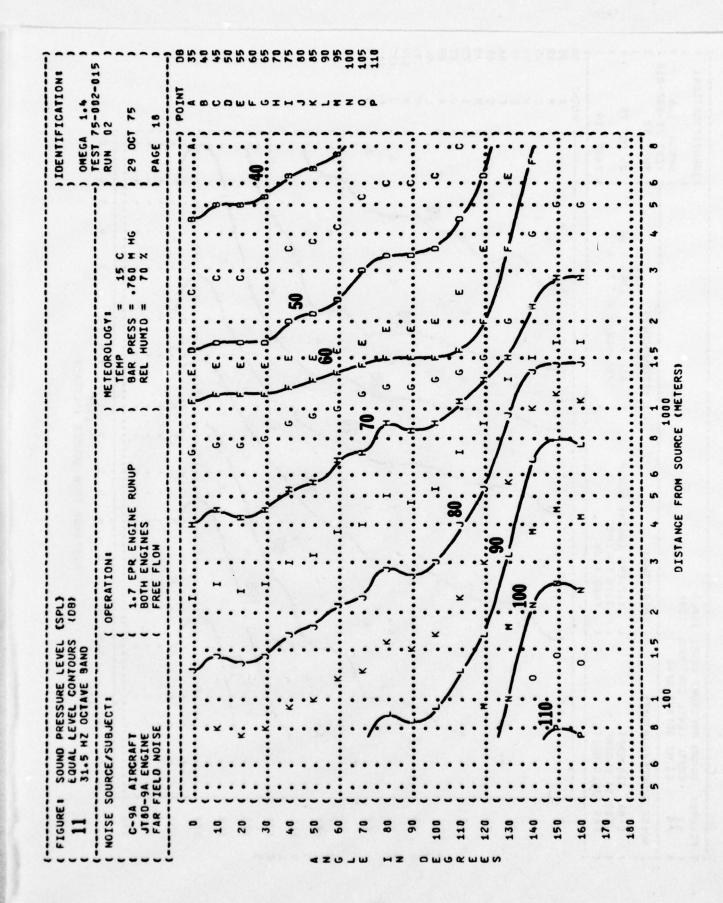


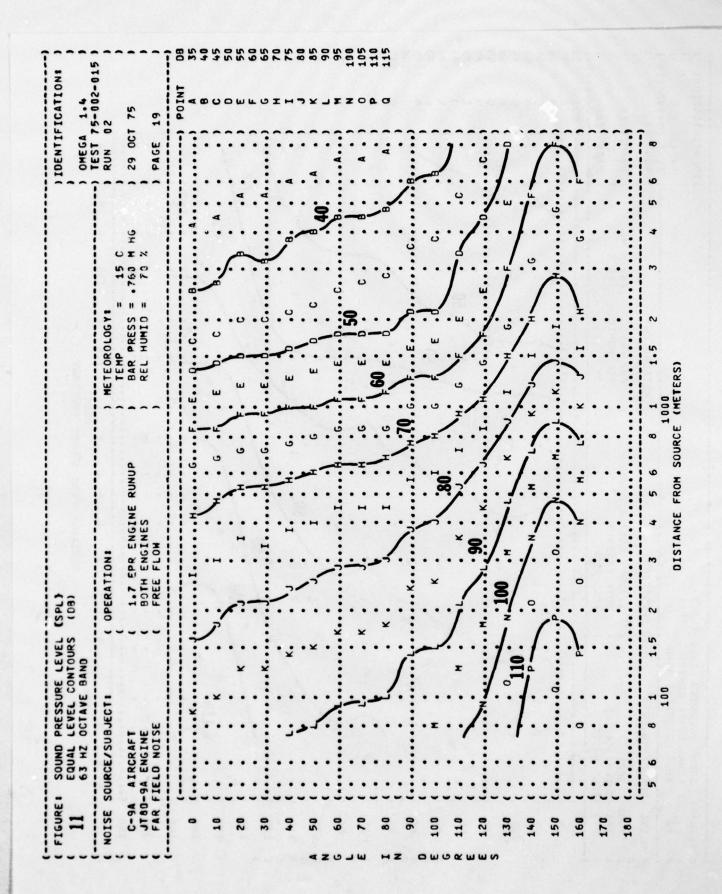


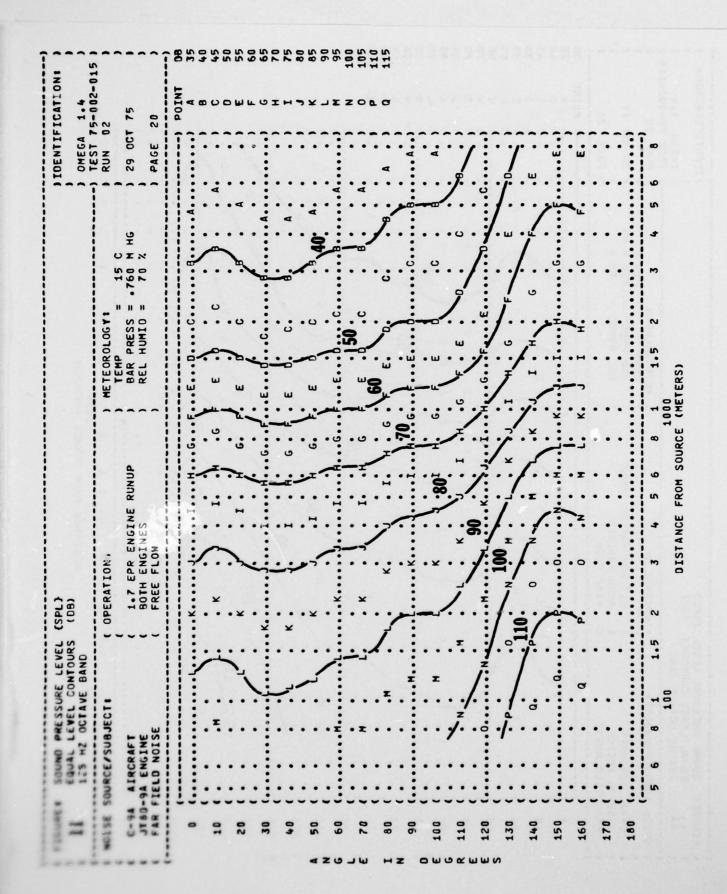


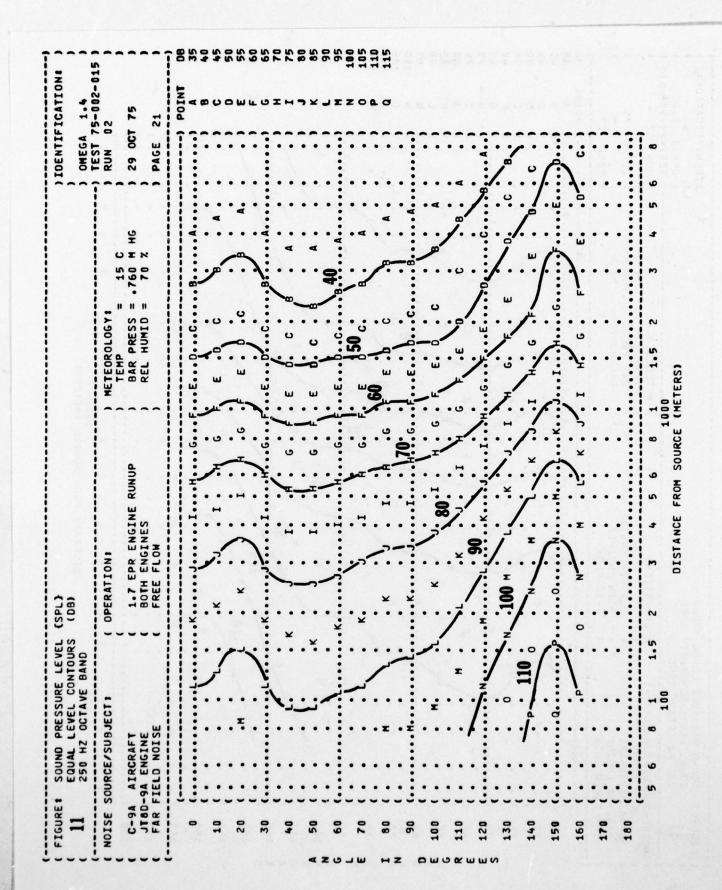
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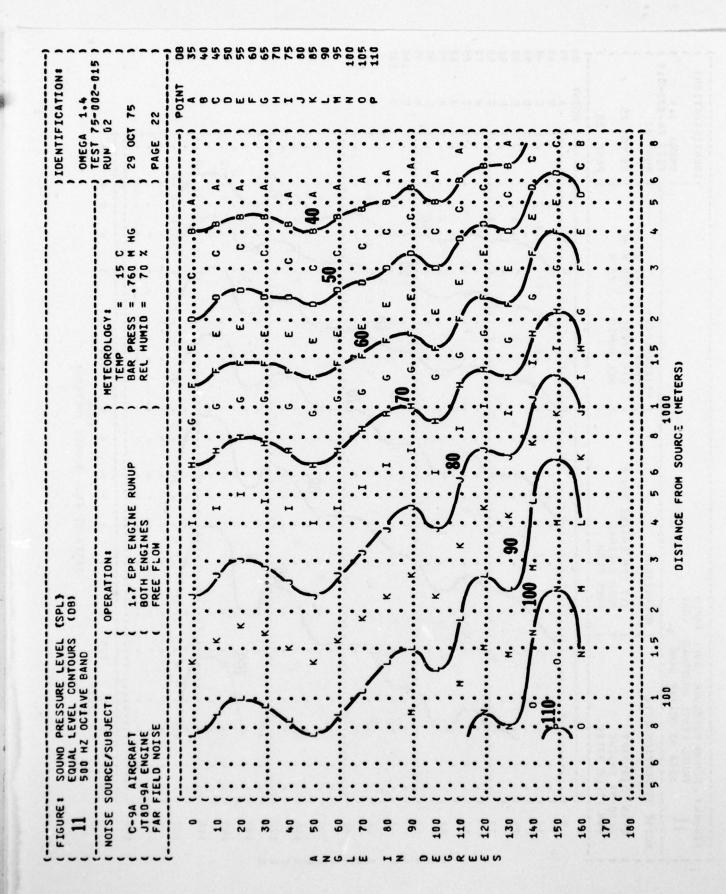
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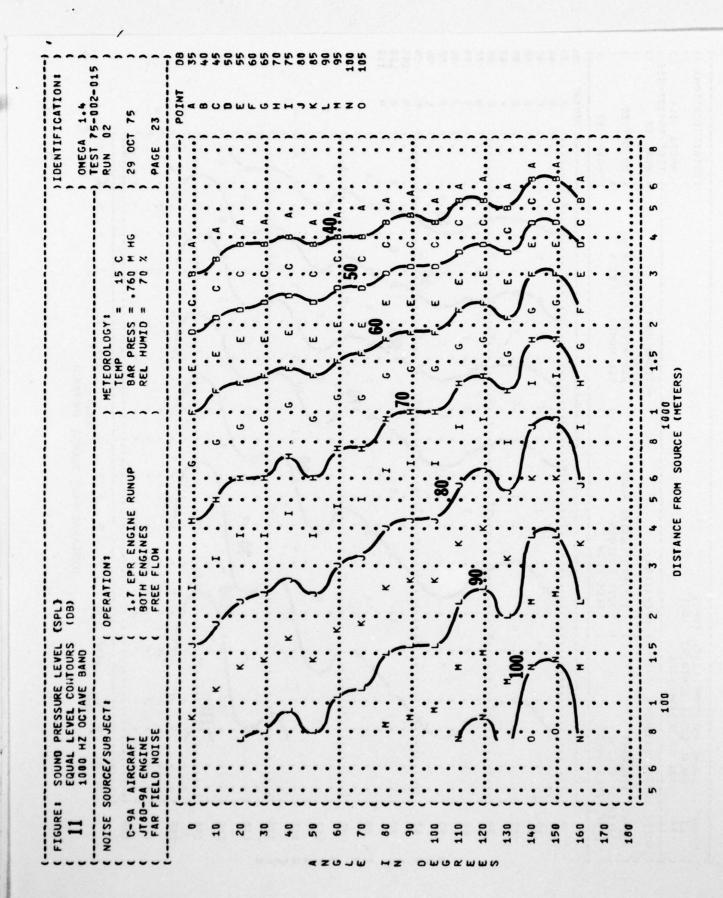


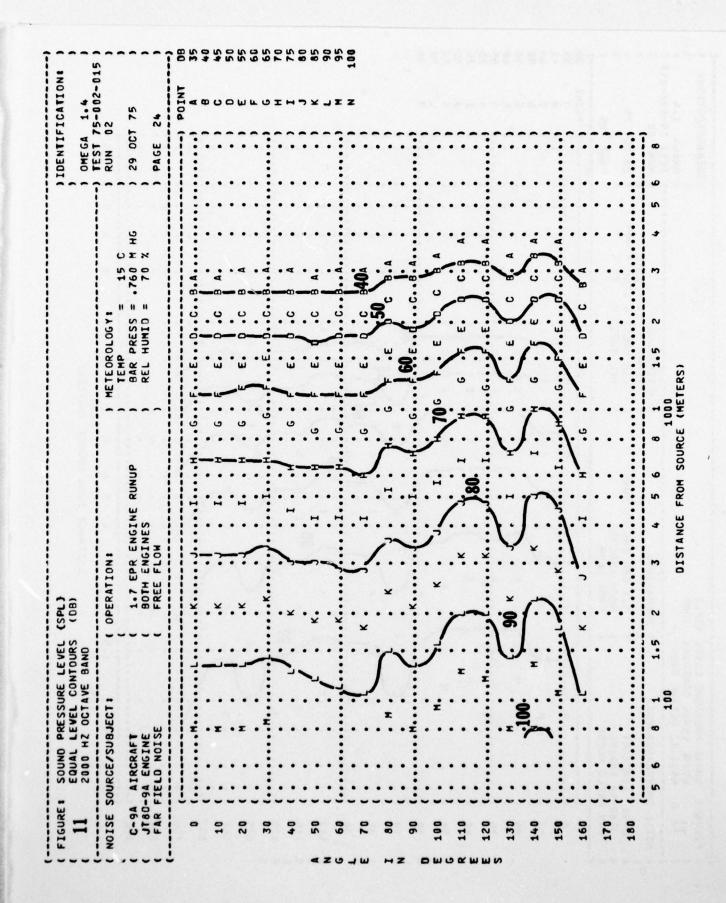








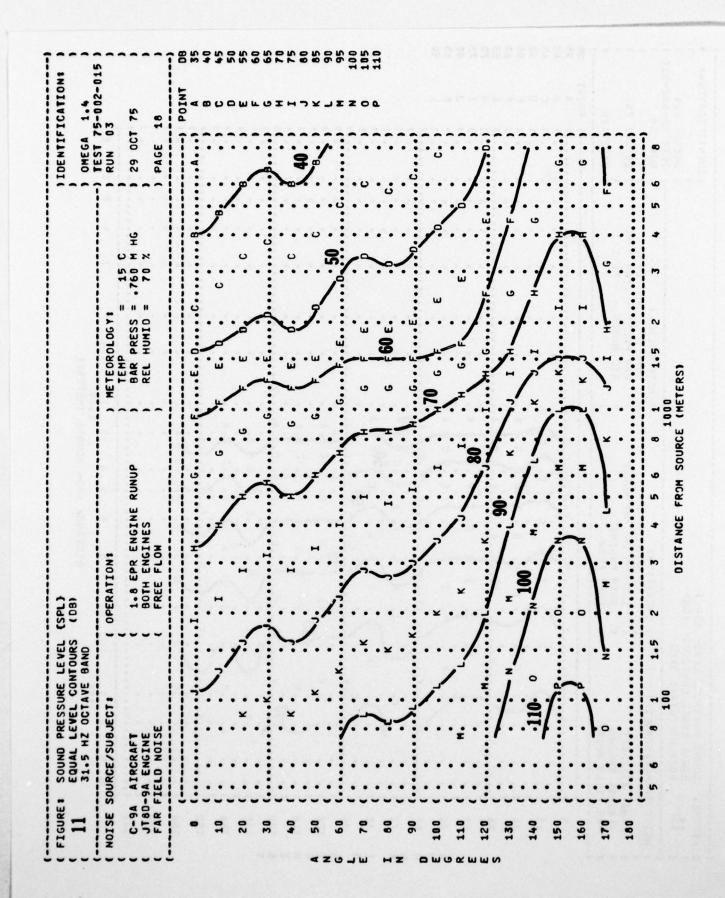


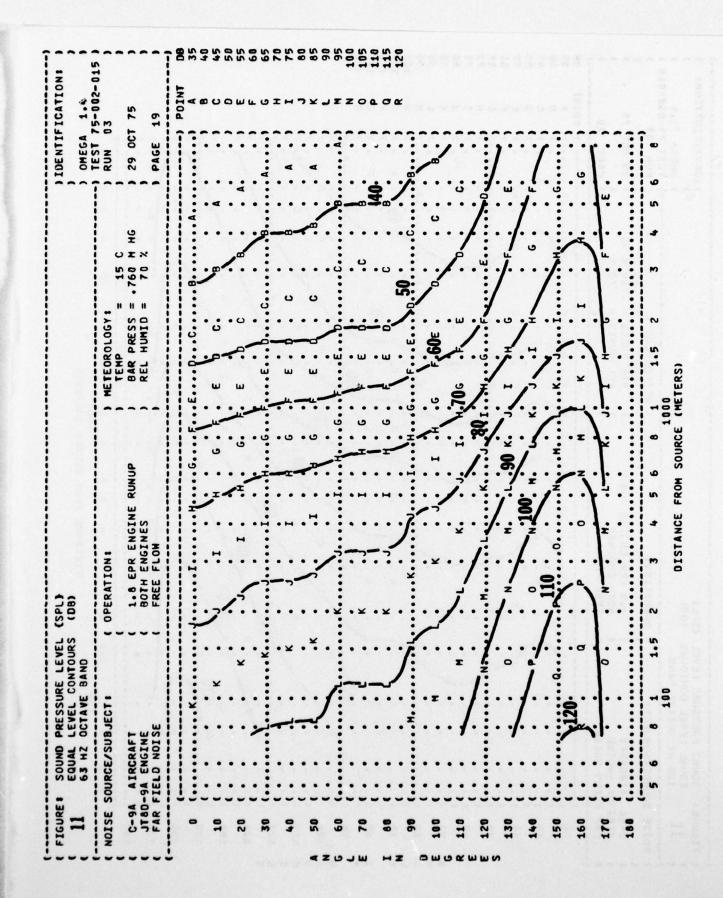


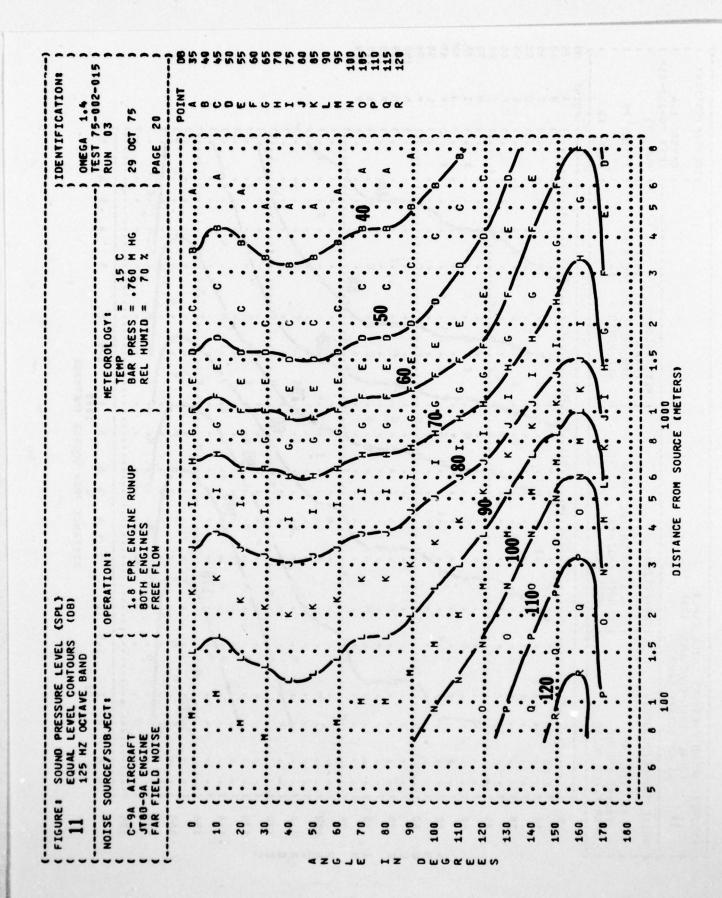
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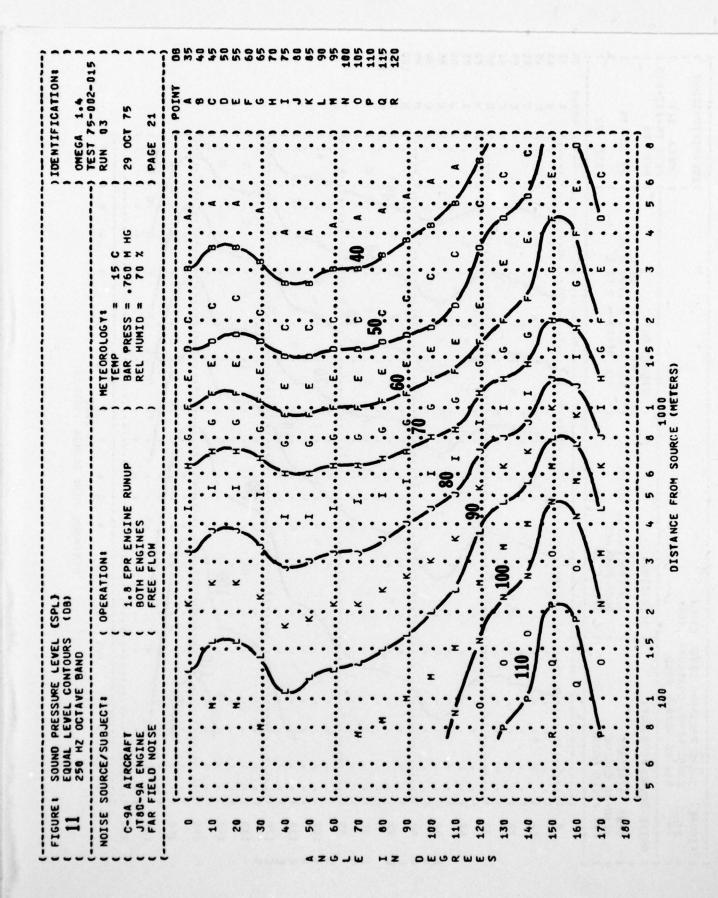
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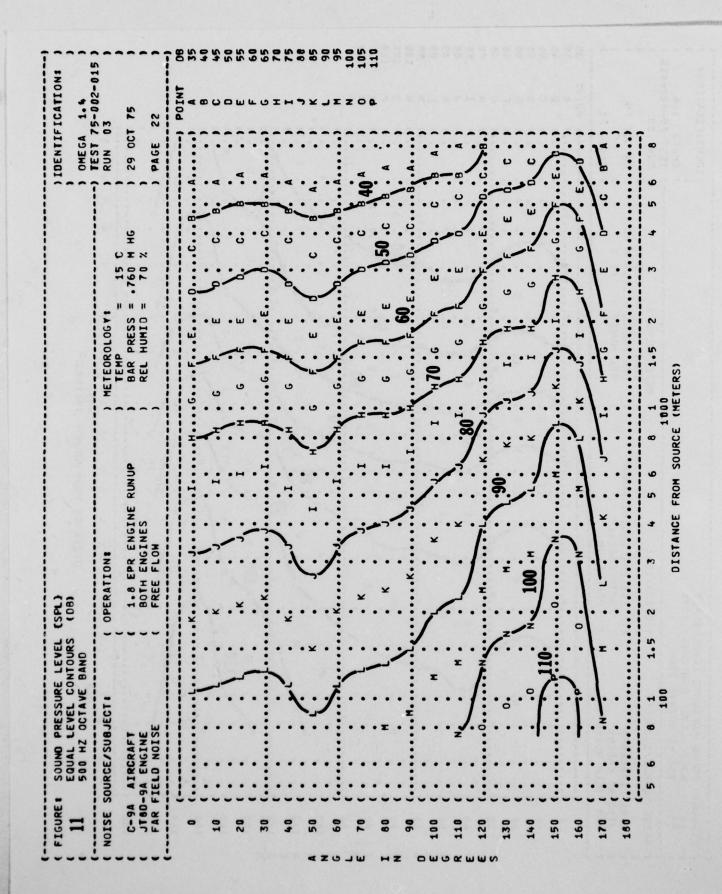
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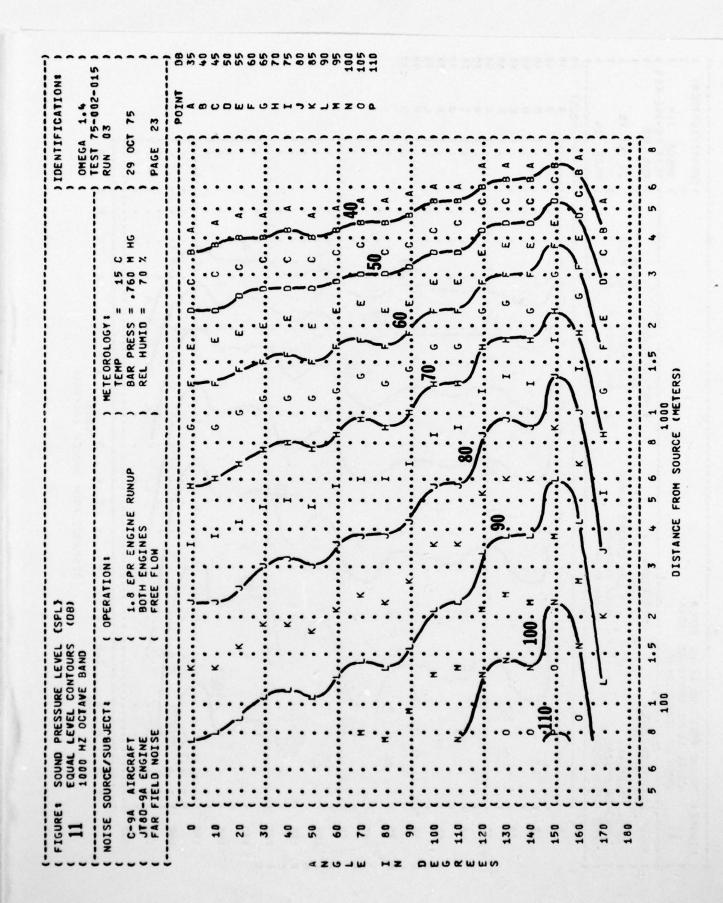


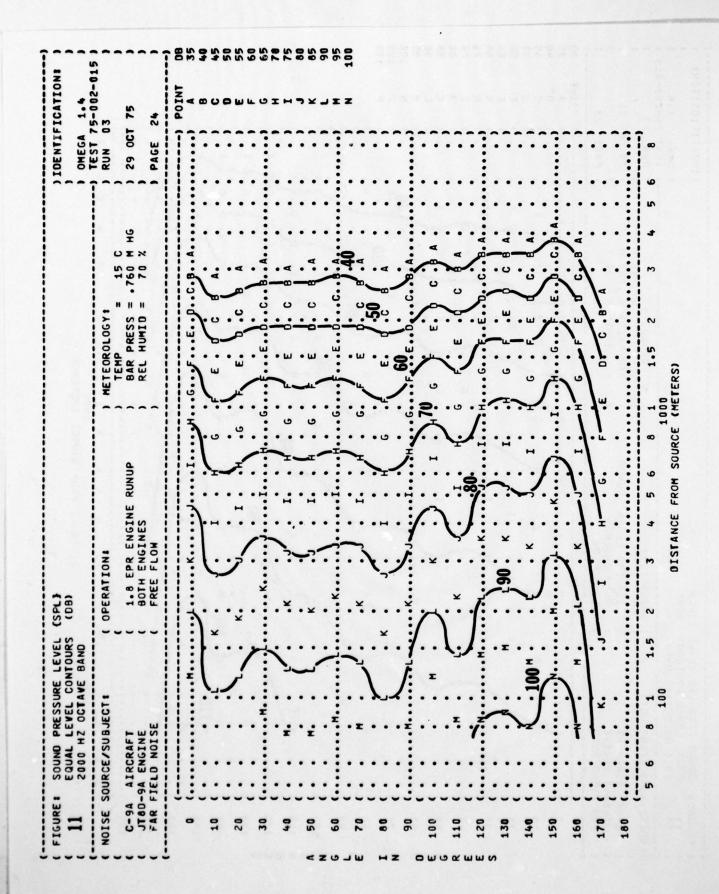












AEROSPACE MEDICAL RESEARCH LAB WRIGHT-PATTERSON AFB OHIO F/G 20/1
USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK: VOLUME 84. C-9A AIRC--ETC(U)
APR 77 R G POWELL
AMRL-TR-75-50-VOL-84 AD-A048 838 UNCLASSIFIED

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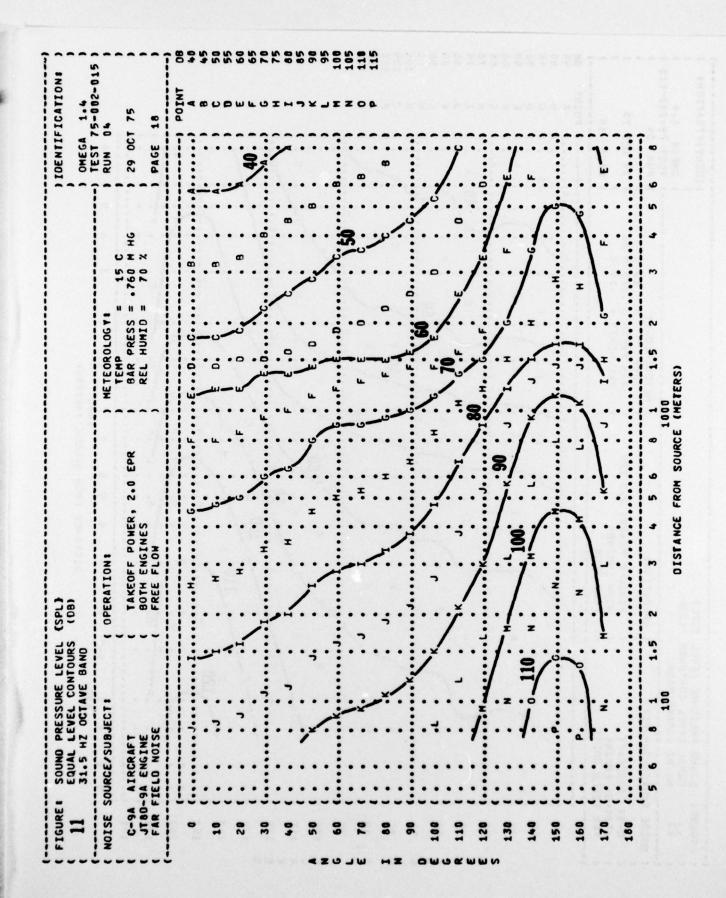


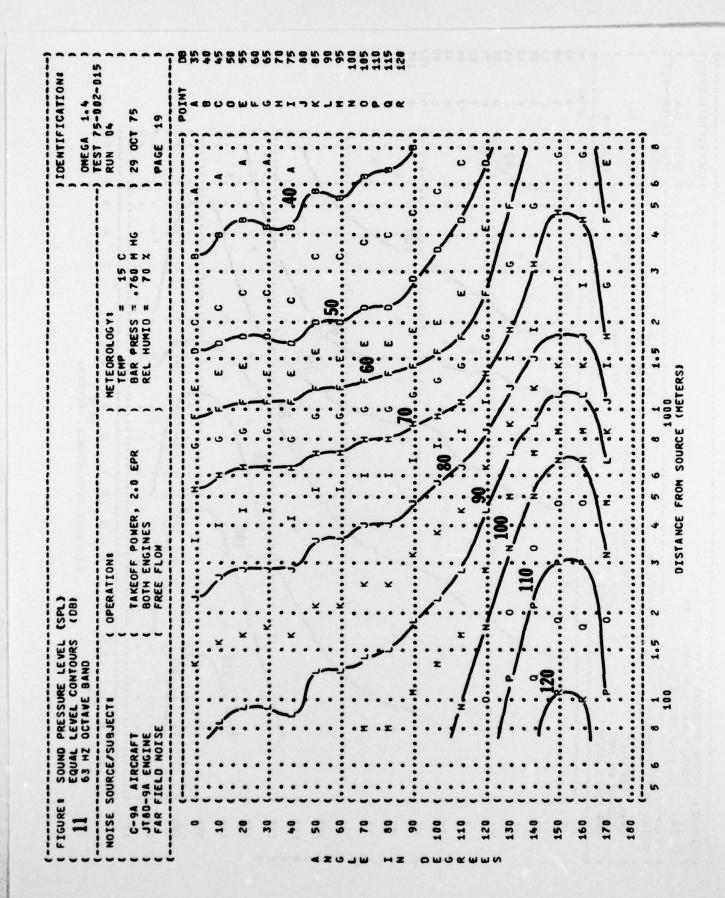


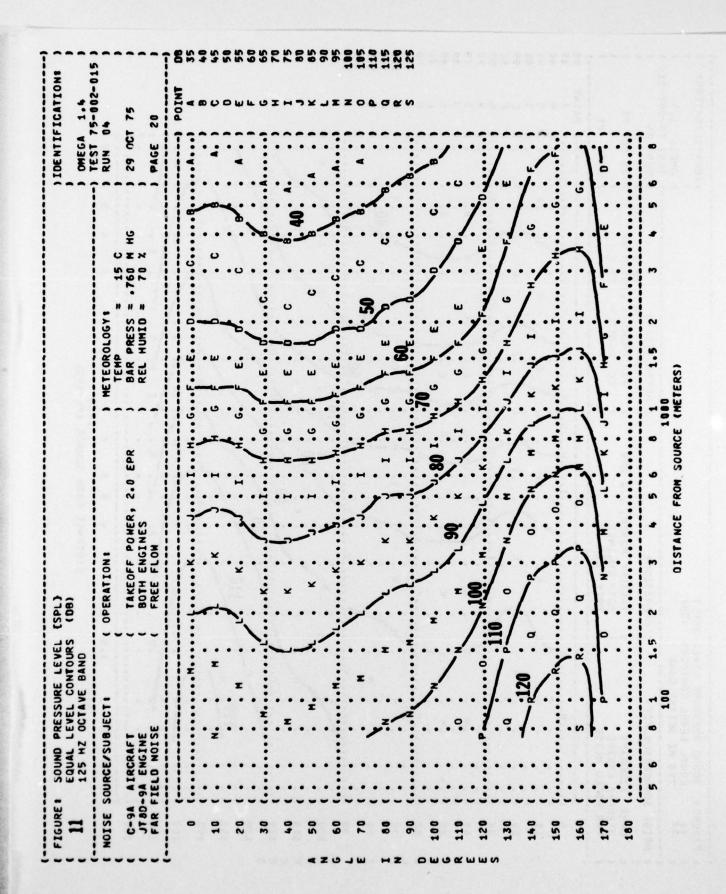


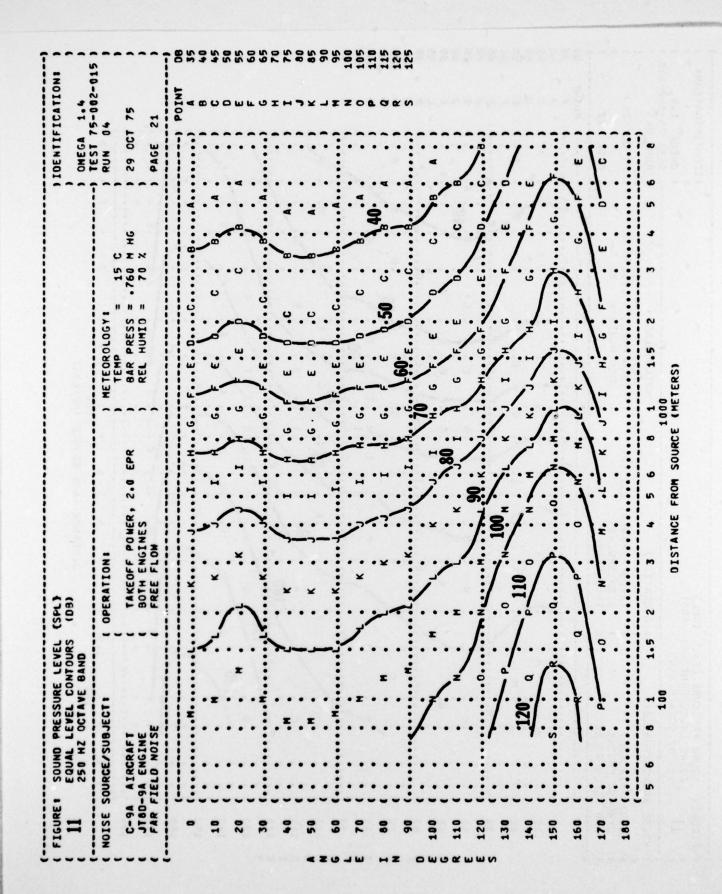
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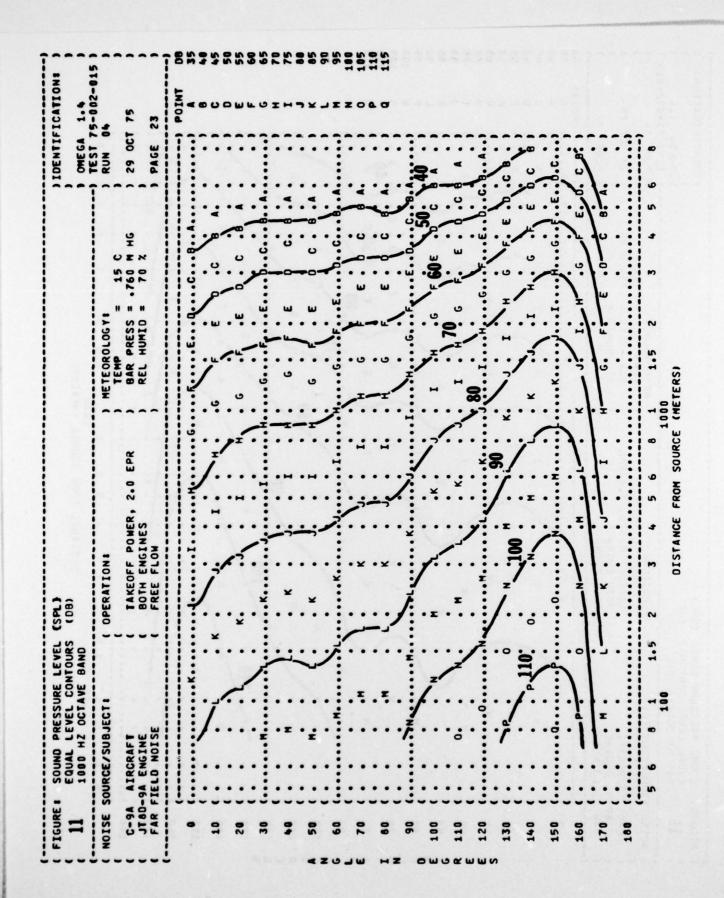


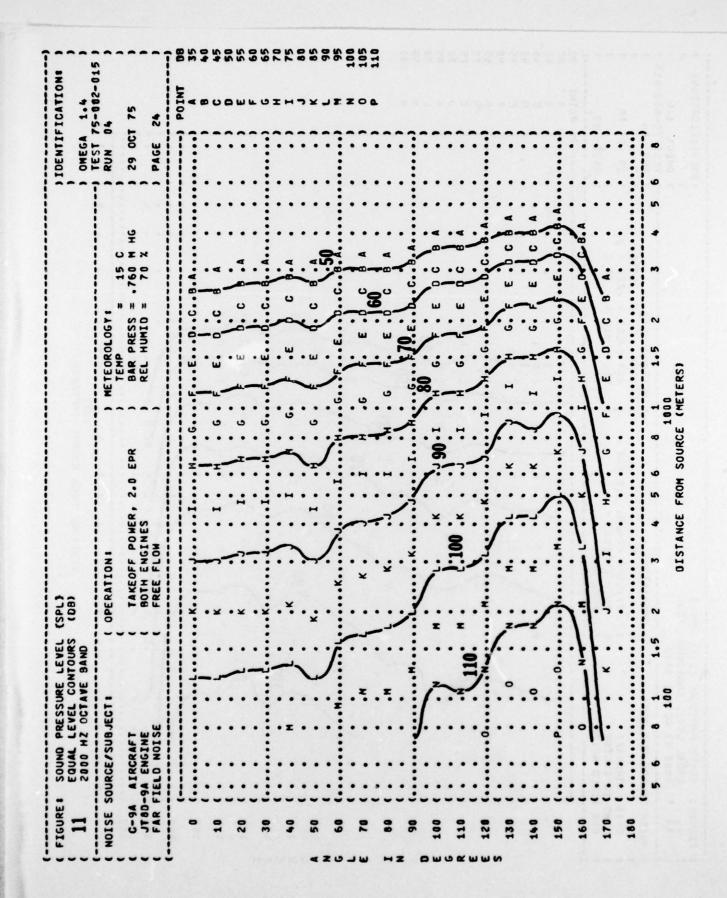


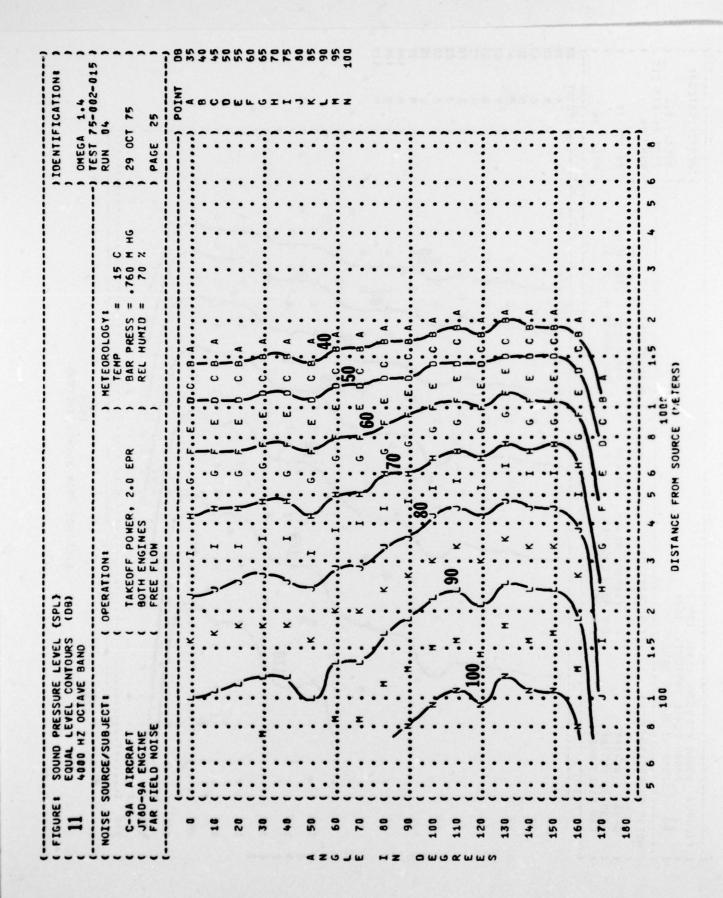




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